

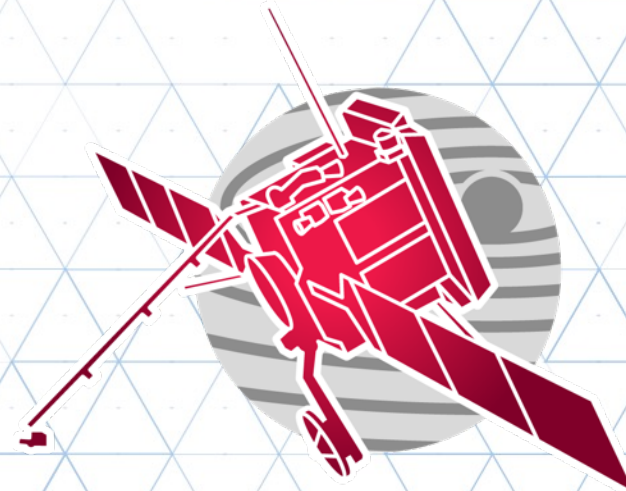
Energetic Particle Measurements with Solar Orbiter

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Javier R.-Podriguez³, and the EPD team

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solar orbiter

Outline

- ❑ **Solar Orbiter (SO) Mission**
- ❑ **Energetic Particle Detector (EPD) Suite on SO**
- ❑ **Questions that will be addressed by SO/EPD:**
 - ❑ **Acceleration**
 - ❑ **Seed Particle**
 - ❑ **Transport**
- ❑ **SO-SIS instrument and data products**
- ❑ **EPD data**

Solar Orbiter Mission

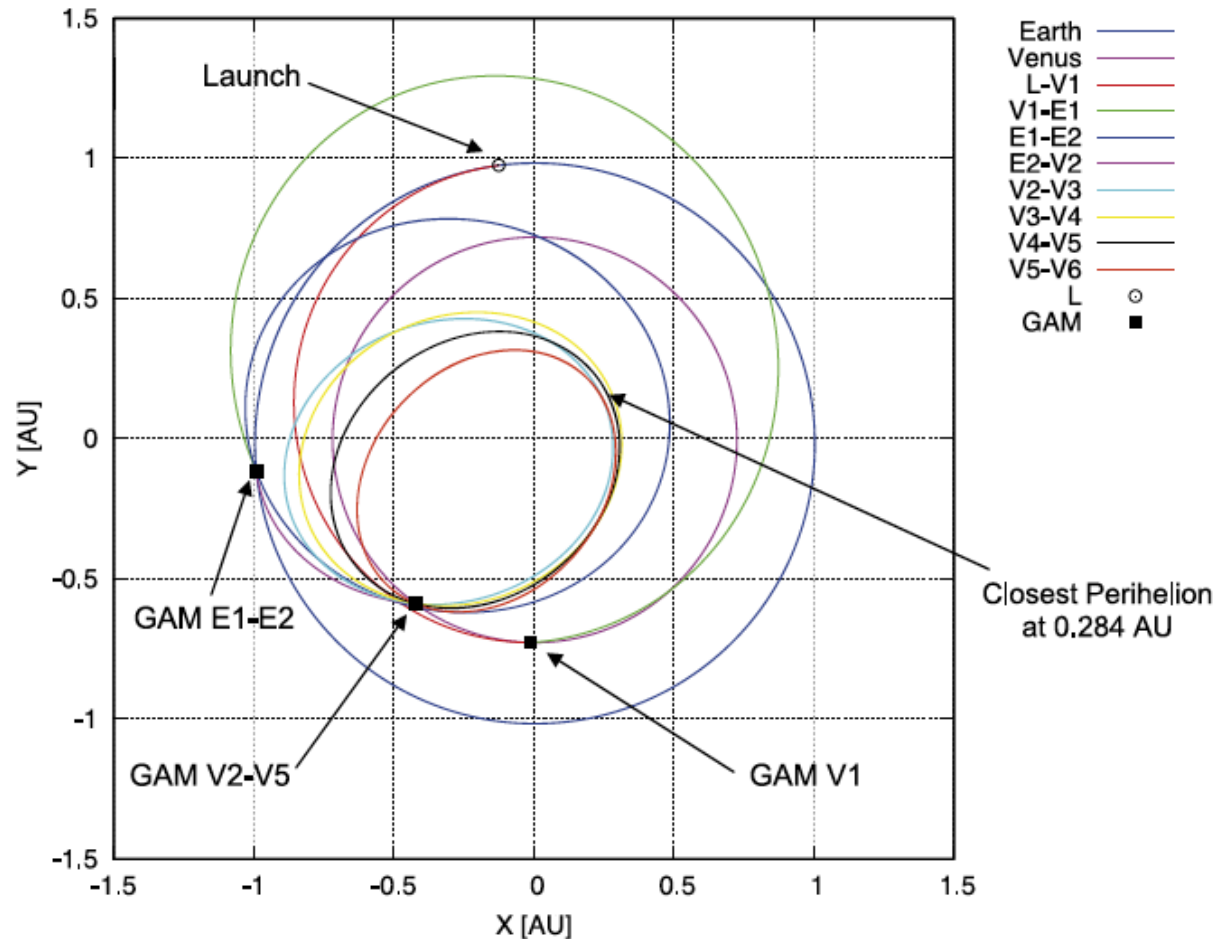


solar orbiter

- A joint ESA-NASA mission to be launched in February 2020
- The spacecraft will take a unique combination of measurements: *in situ* measurements will be used alongside remote sensing close to the Sun to relate these measurements back to their source regions and structures on the Sun's surface.
- Solar Orbiter will set about answering four top-level science questions:
 - ***What drives the solar wind and where does the coronal magnetic field originate from?***
 - ***How do solar transients drive heliospheric variability?***
 - ***How do solar eruptions produce energetic particle radiation that fills the heliosphere?***
 - ***How does the solar dynamo work and drive connections between the Sun and the heliosphere?***

Müller et al. (2013)

Mission Design



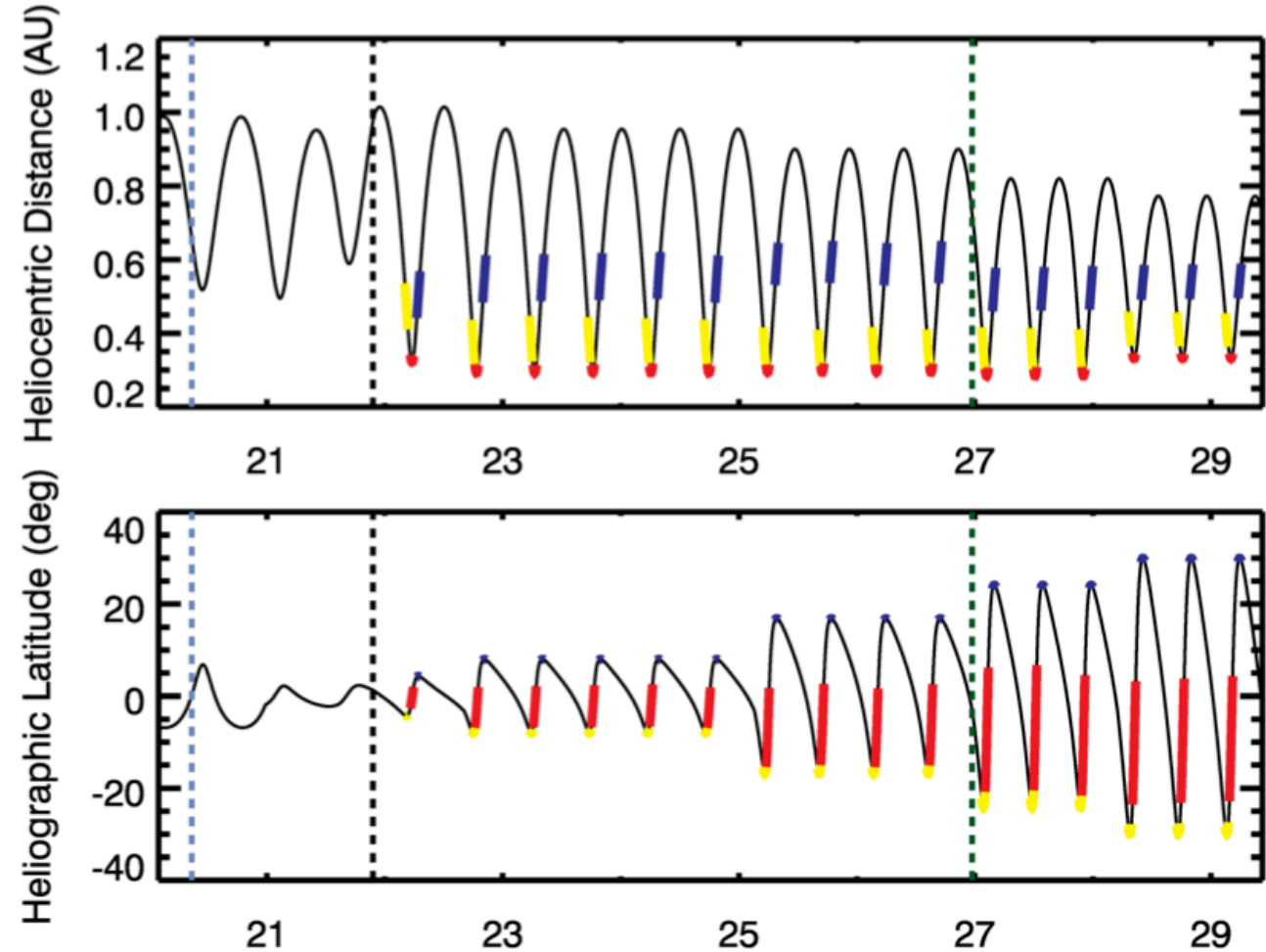
- **NASA-provided launcher**
- **First perihelion is reached 1.5 years after launch**
- **Operating orbit has a 168 days period**
- **Resonant orbit with Venus**
- **Minimum perihelion at 0.28 AU**
- **7 years nominal mission at 25° inclination**

Figure 2 *Solar Orbiter's* trajectory viewed from above the ecliptic (January 2017 launch). The gravity assist maneuvers (GAM) at Earth (E) and Venus (V) are indicated, along with the orbits of these two planets.

Mission phases (Feb 2020 launch)

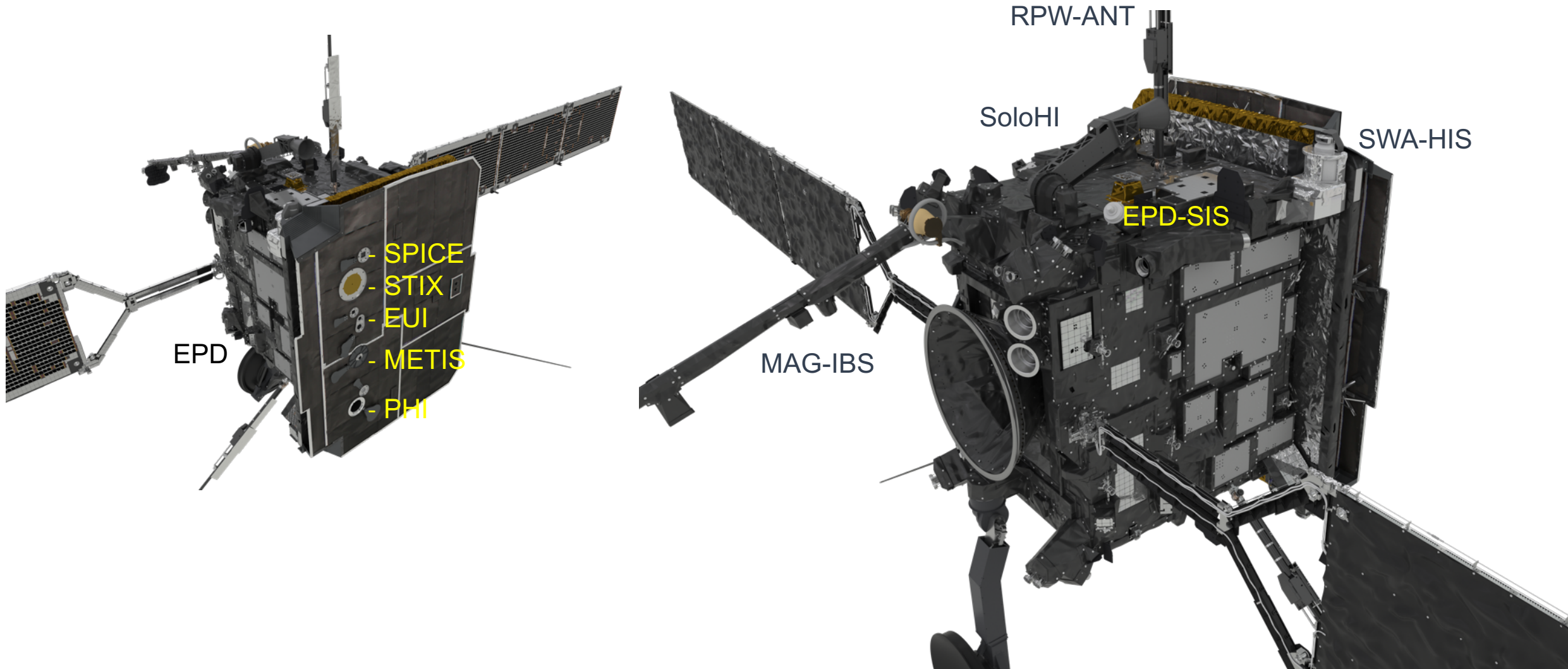


- **CP** Cruise Phase
 - From 97 days after launch until the planetary fly-by that puts the spacecraft into the NMP orbit.
 - Short: 561 days
- **NMP** Nominal Mission Phase.
 - for Feb 2020 launch, this starts in Nov 2021



Solar Orbiter Spacecraft

10 instrument suite: 4 in-situ; 6 remote



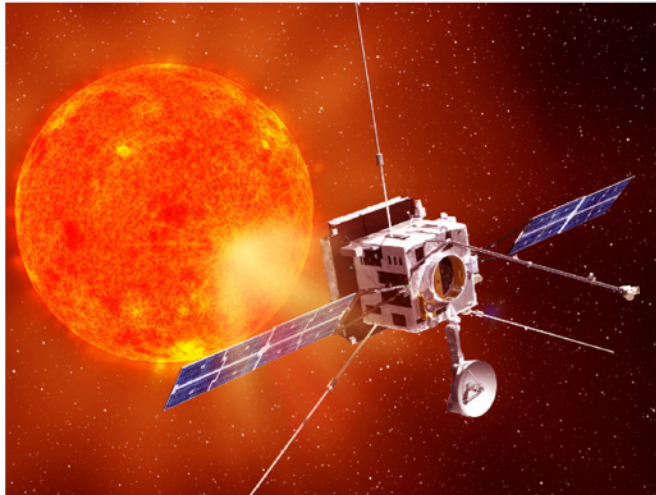
Solar Orbiter: Energetic Particle Science Question



ESA/SRE(2011)14
July 2011

Solar Orbiter

Exploring the Sun-heliosphere
connection



Definition Study Report

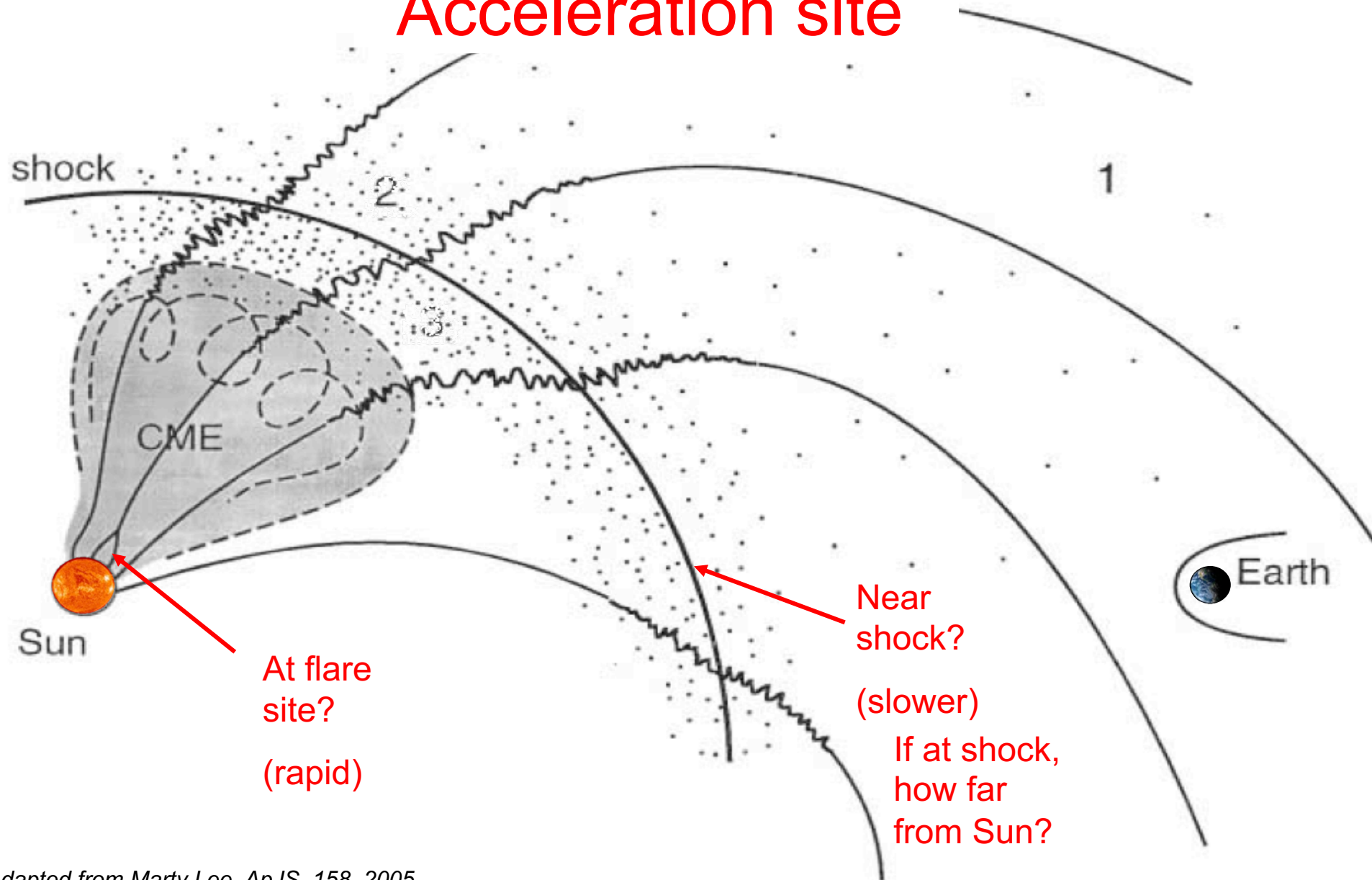
How do solar eruptions produce energetic particle radiation that fills the heliosphere?

1) Acceleration

2) Injection/Seed population

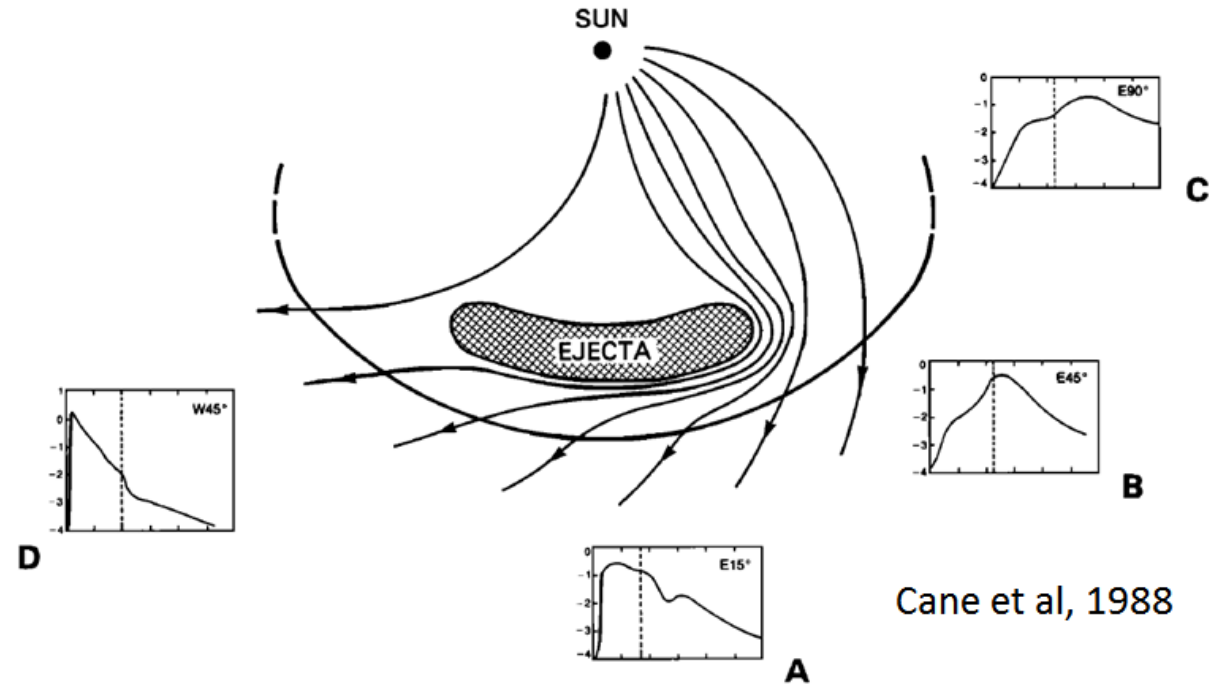
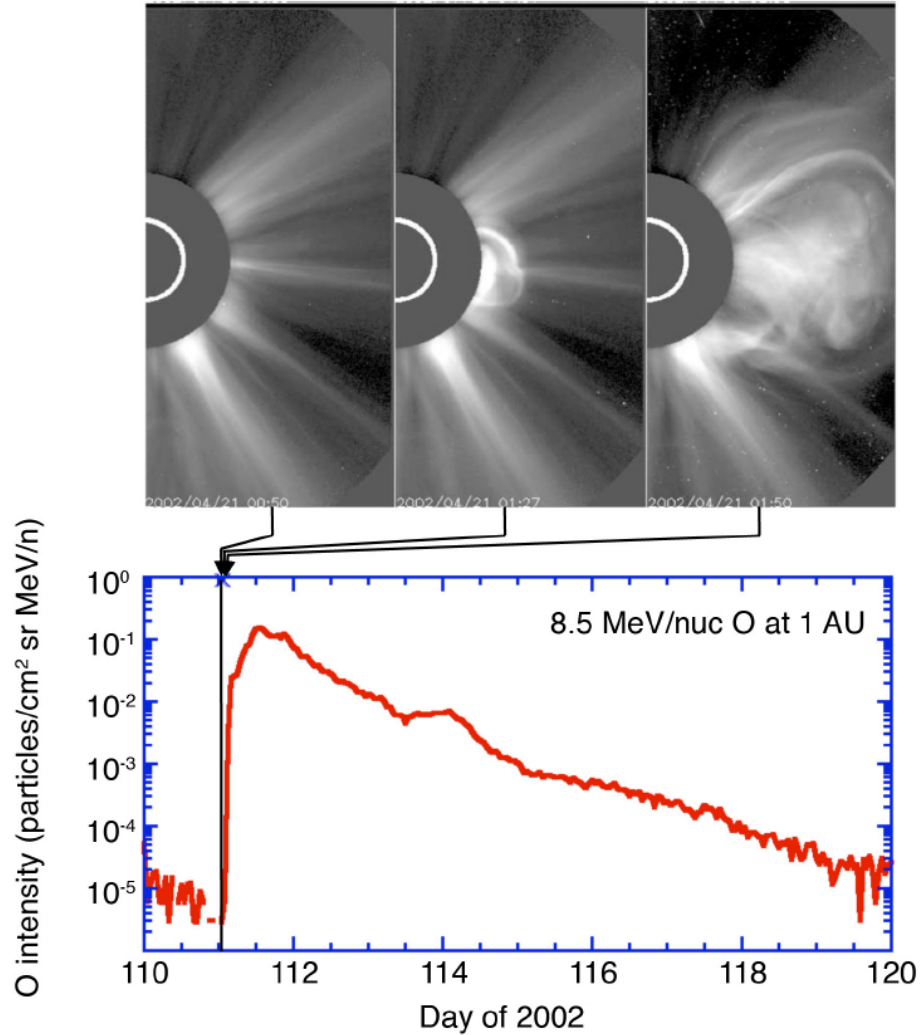
3) Transport seed population

Acceleration site



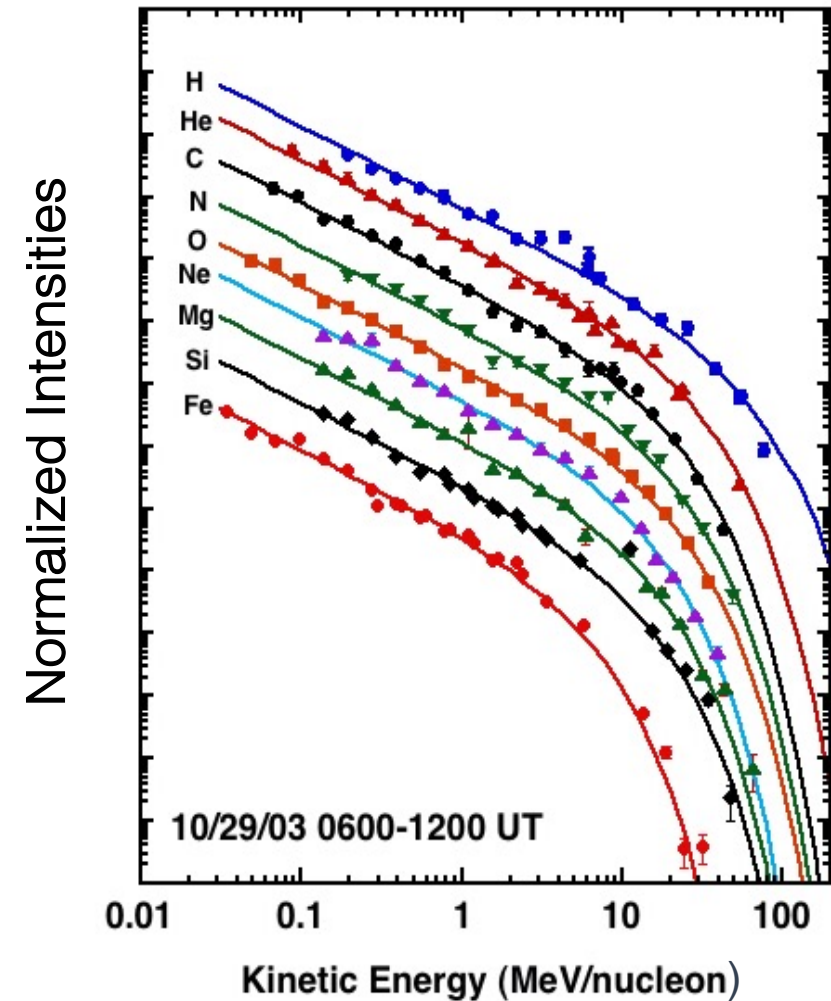
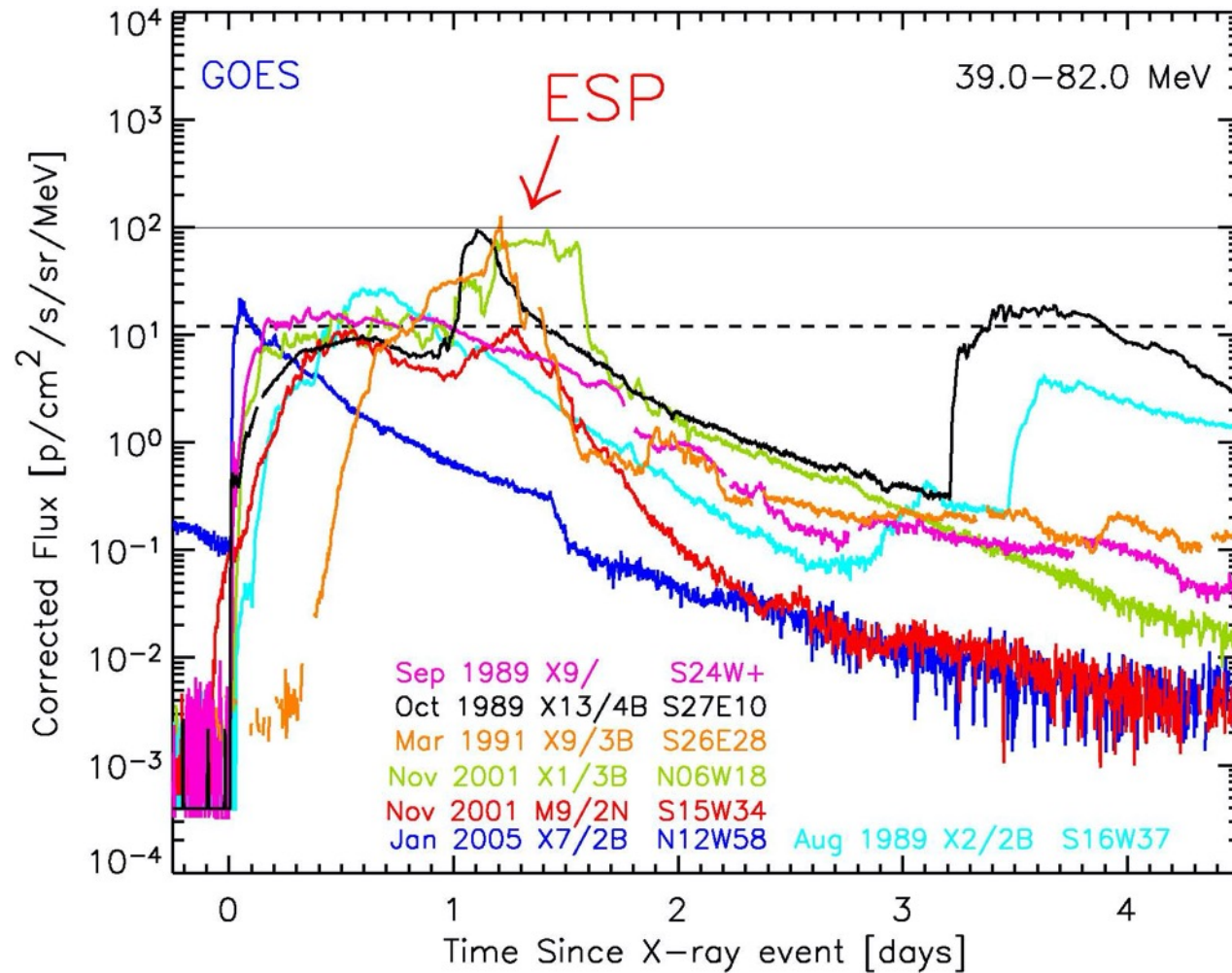
Adapted from Marty Lee, *ApJS*, 158, 2005

1. Shock Acceleration



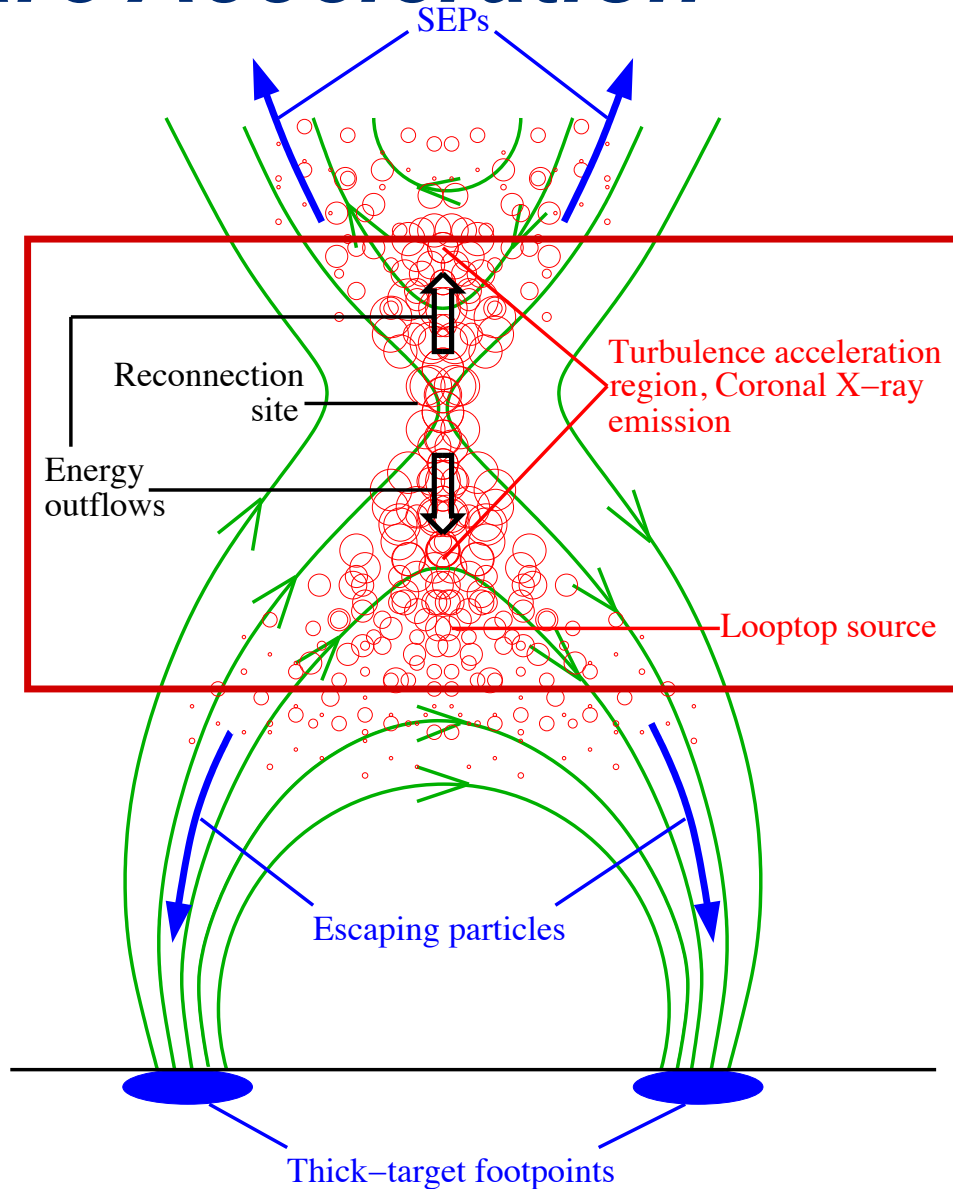
Cane et al, 1988

Evidence for Shock Acceleration

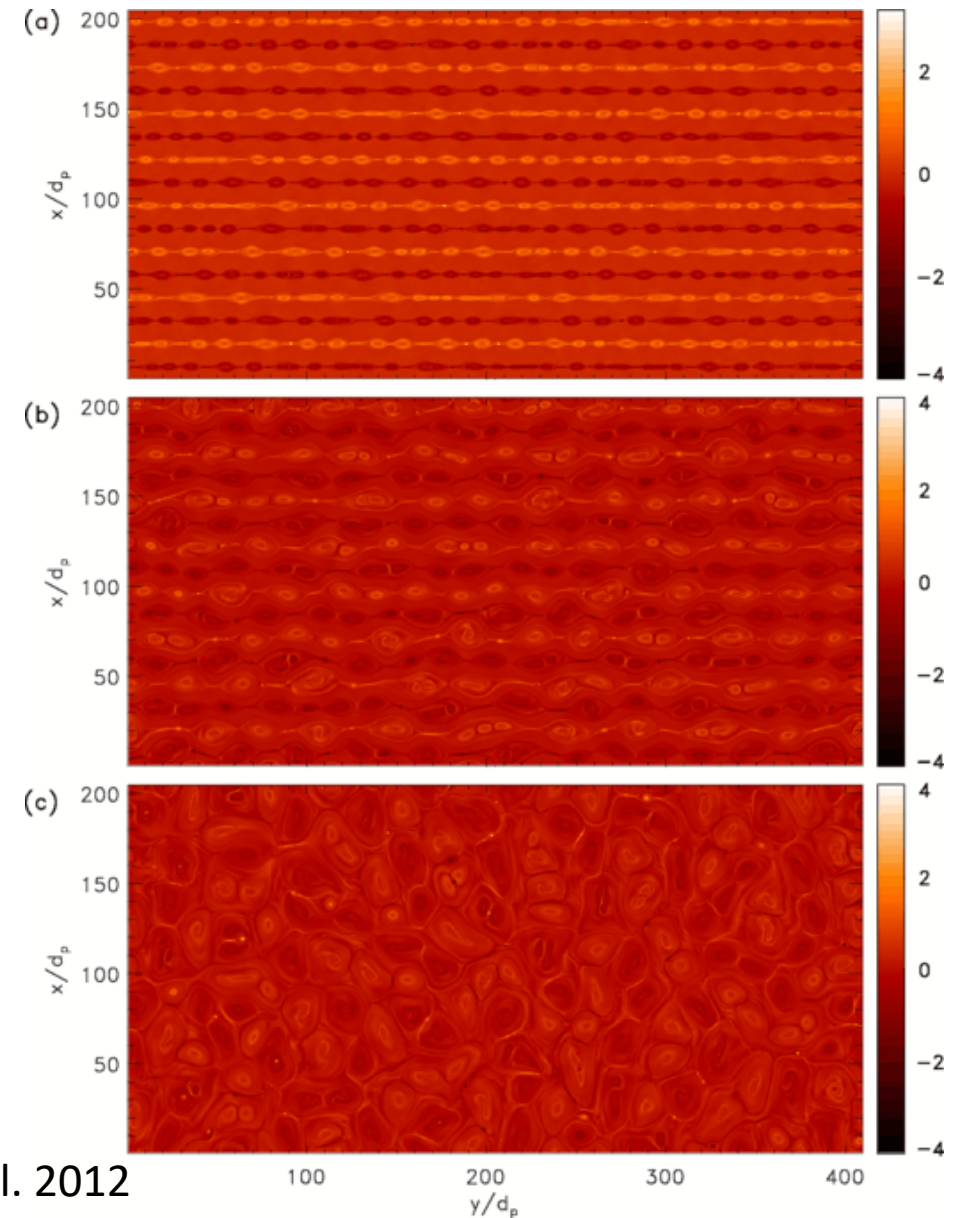


Mewaldt et al. 2005

2. Flare Acceleration

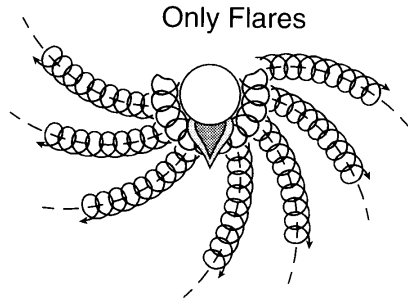


Liu et al. 2013



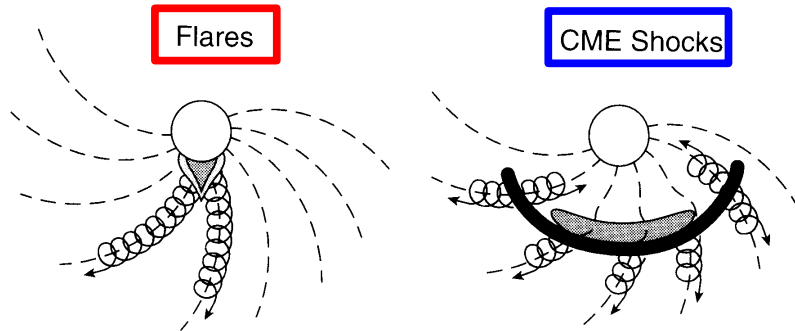
Drake et al. 2012

Old Picture:

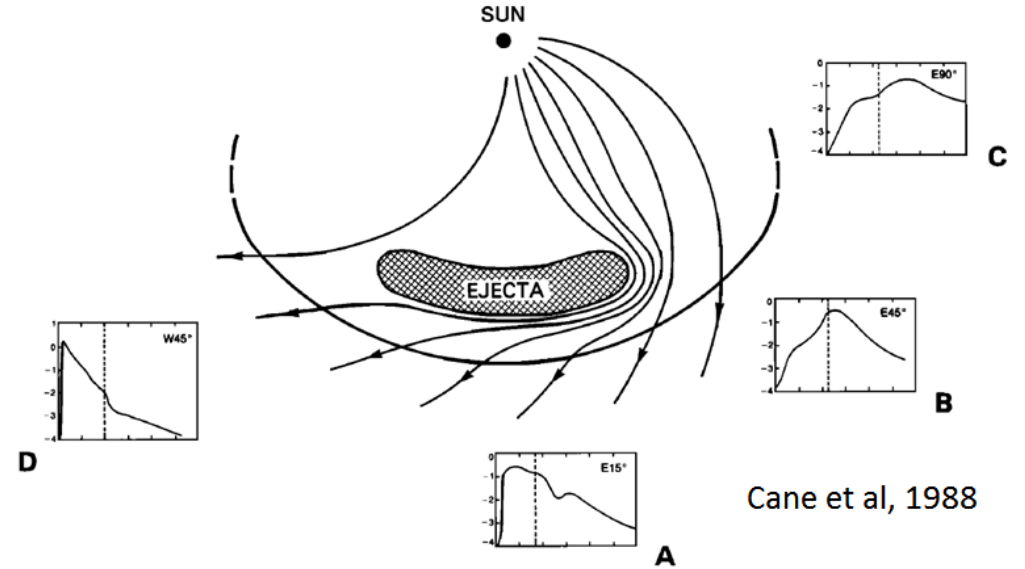
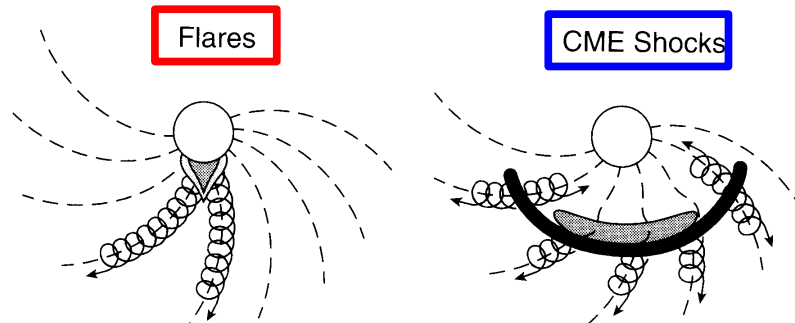


CME-driven shocks play an important role in accelerating particles But....

~~Old~~ new
New Picture:



New Picture:



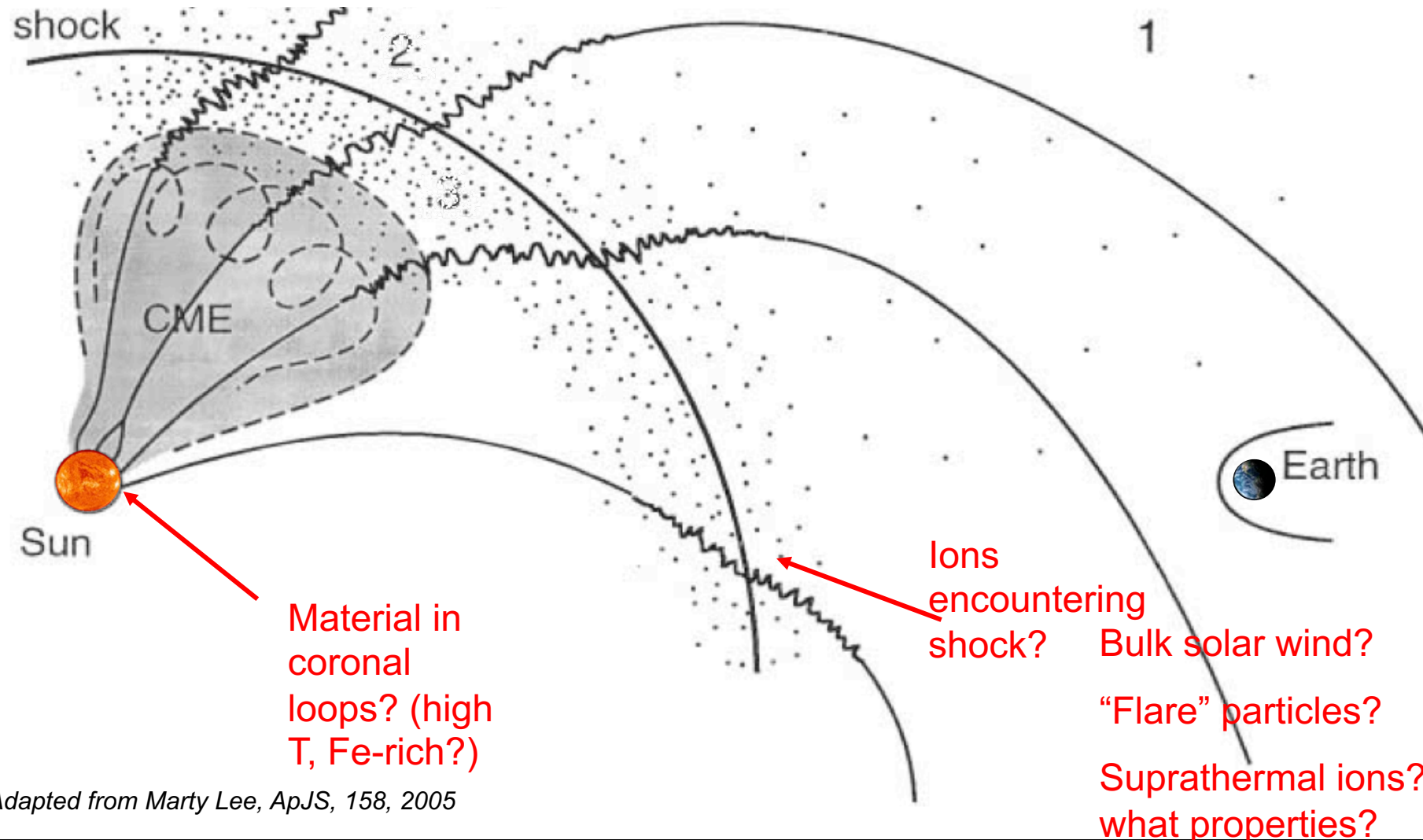
Cane et al, 1988

**Life is more complicated:
Flares can contribute to shock-accelerated particles.**

Composition helps to understand how

But it does not solve all problems!

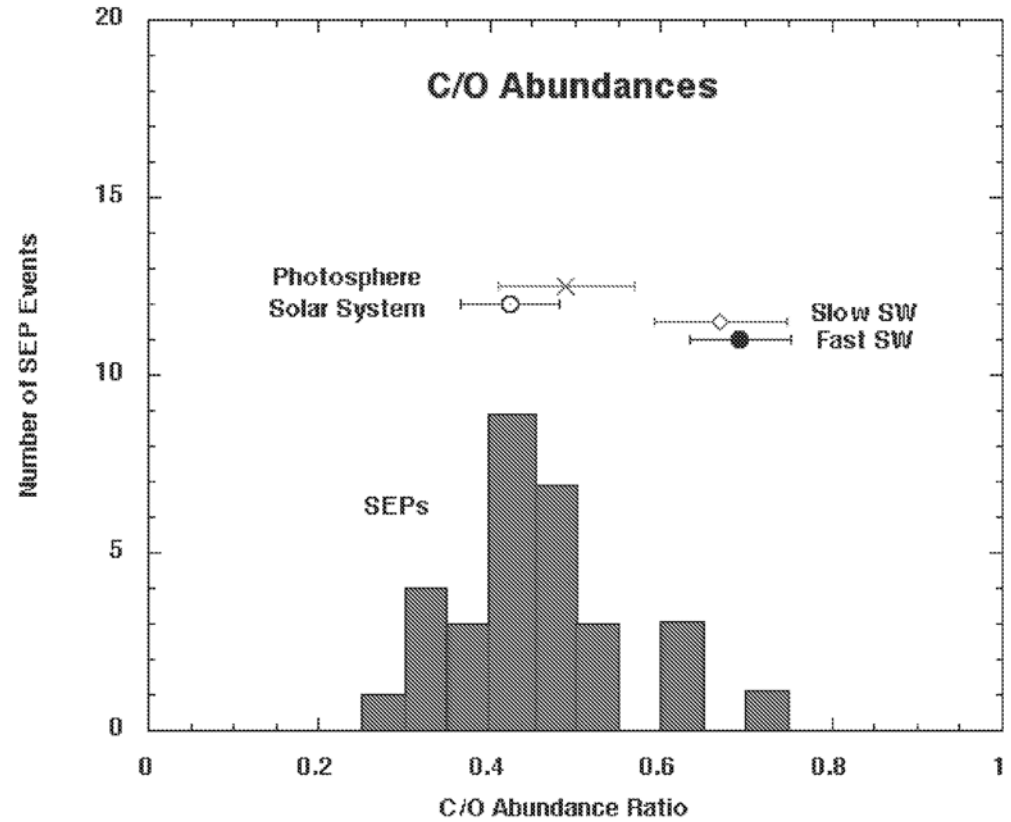
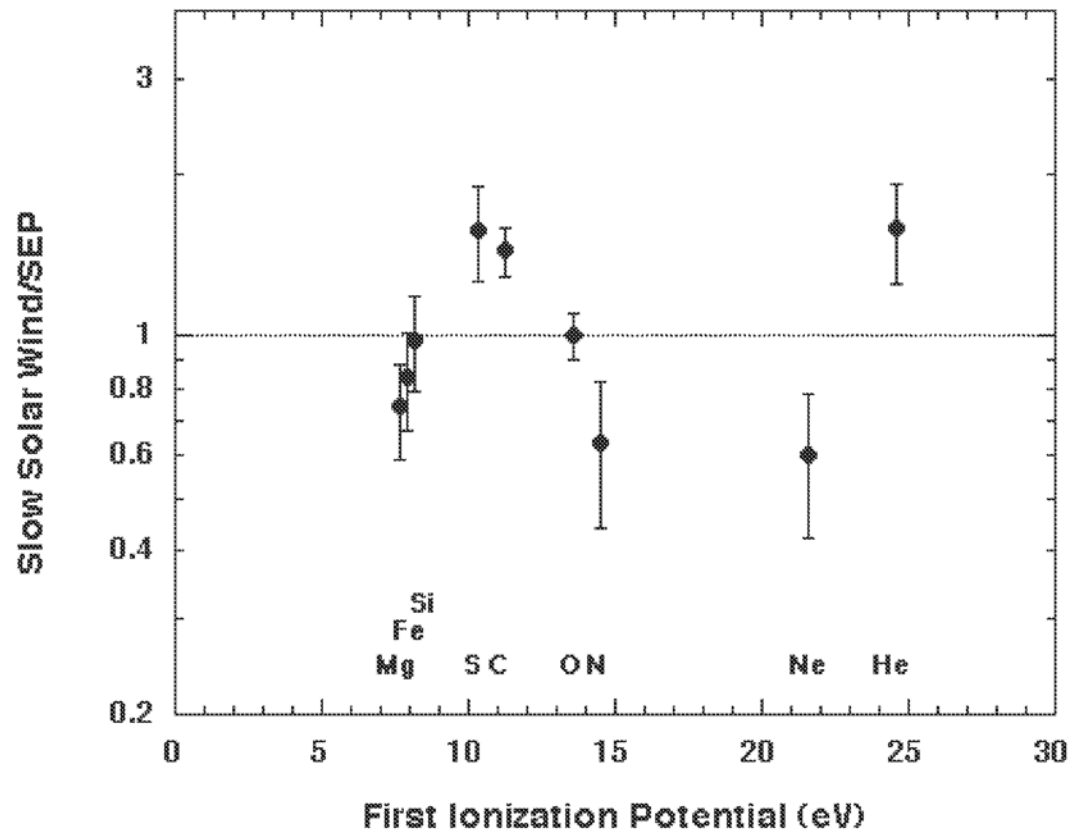
What is being accelerated - seed population?



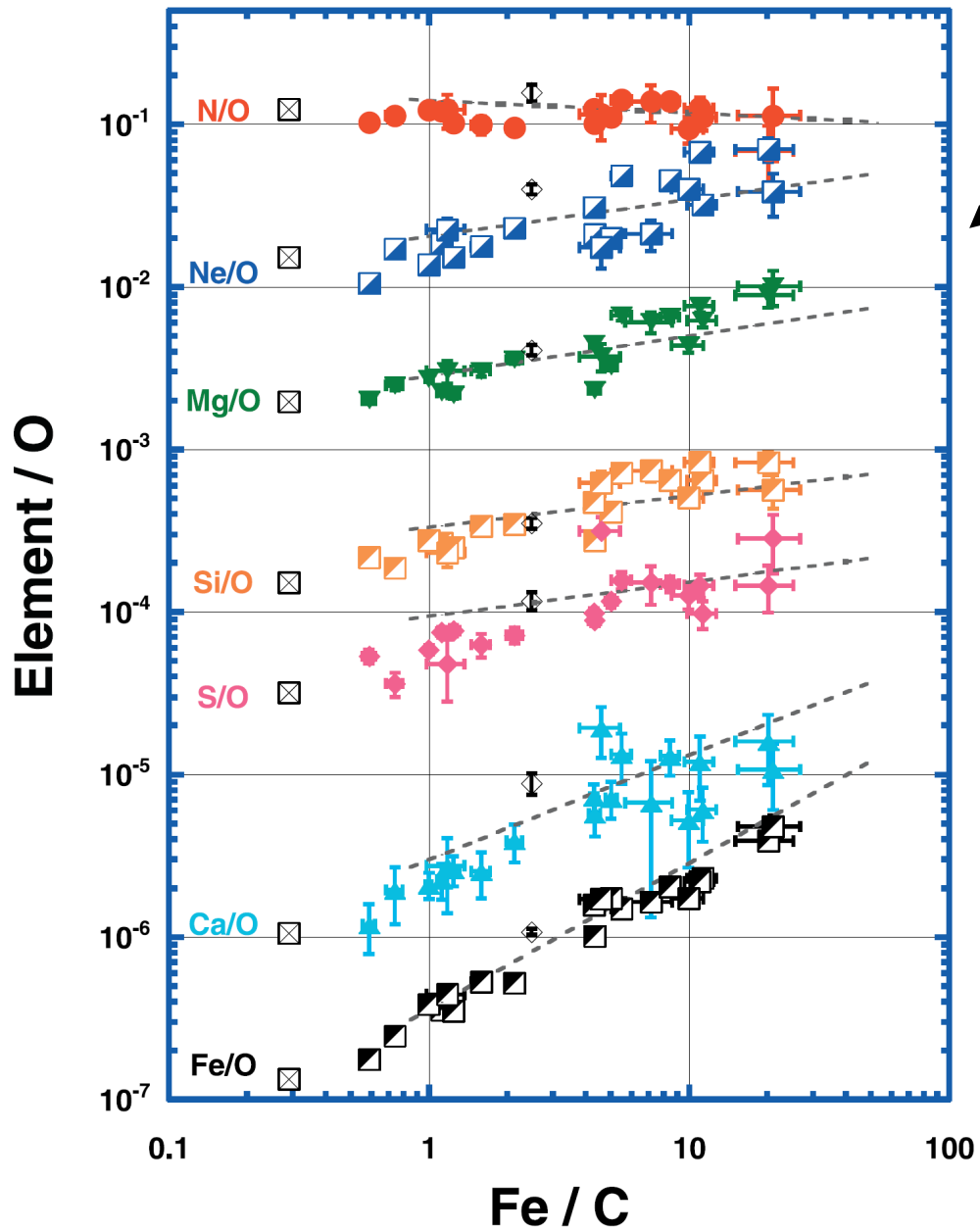
Adapted from Marty Lee, *ApJS*, 158, 2005

What is being accelerated?

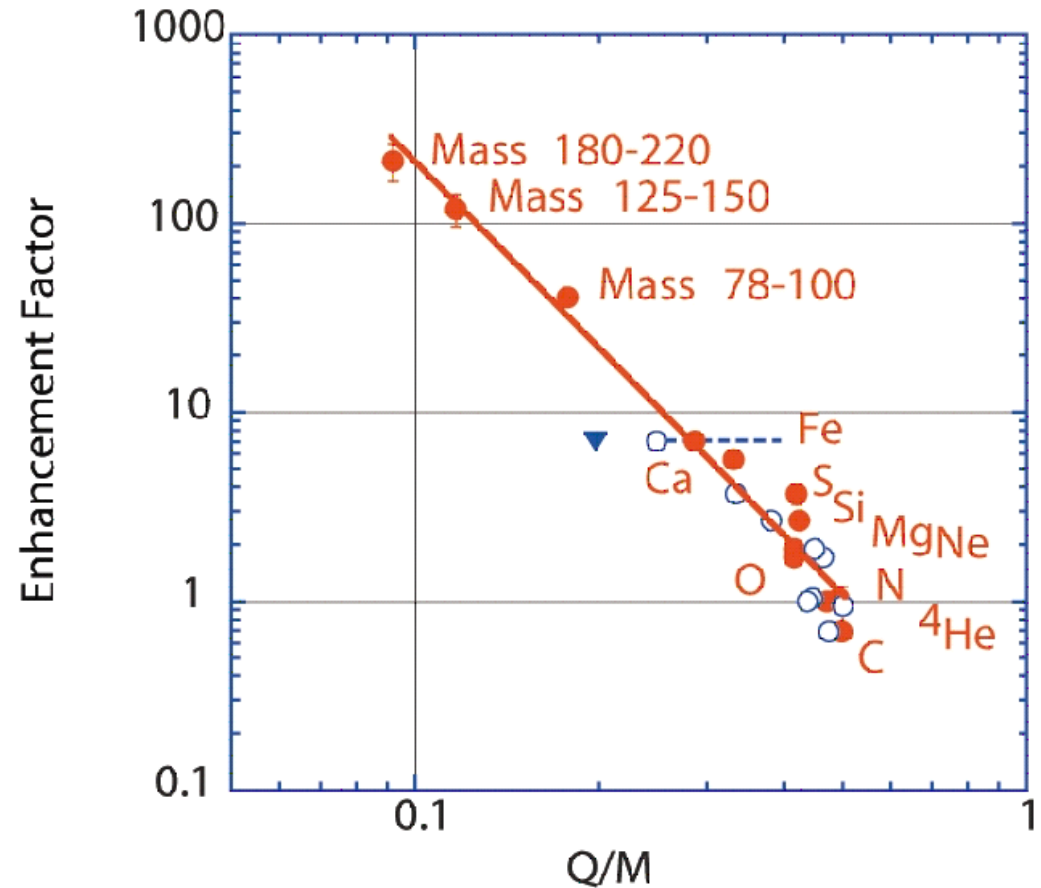
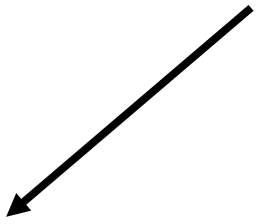
- Solar Wind & SEP have different composition



Mewaldt et al. 2001



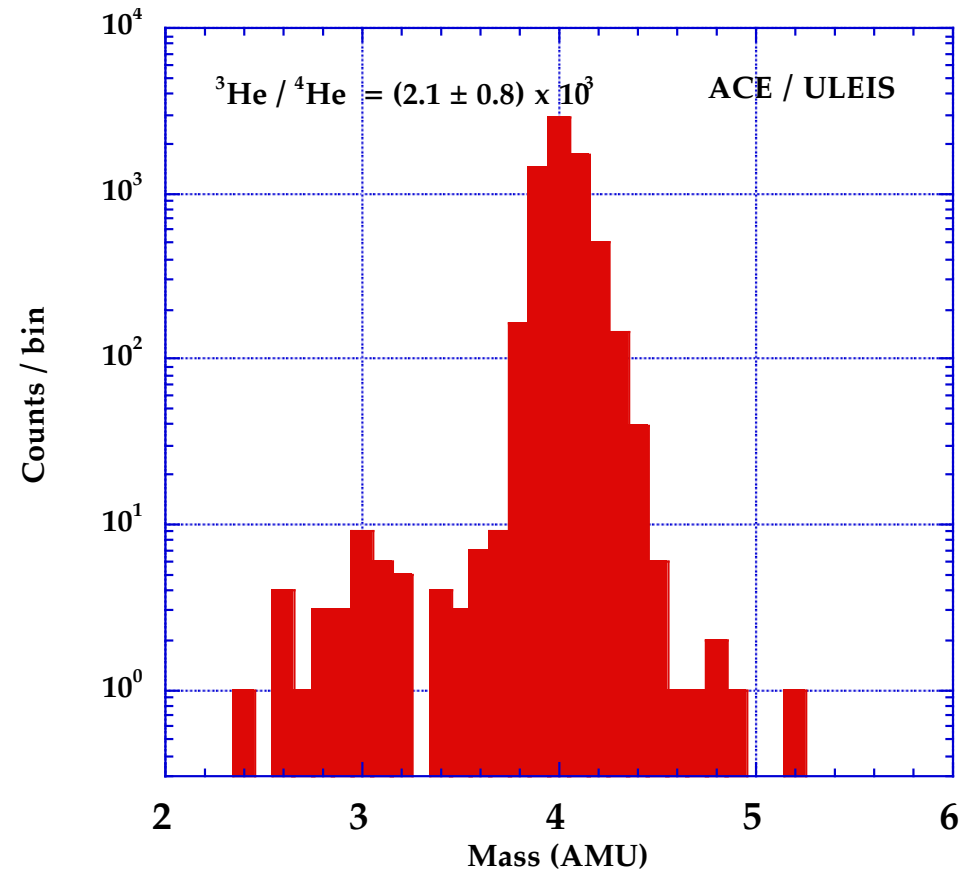
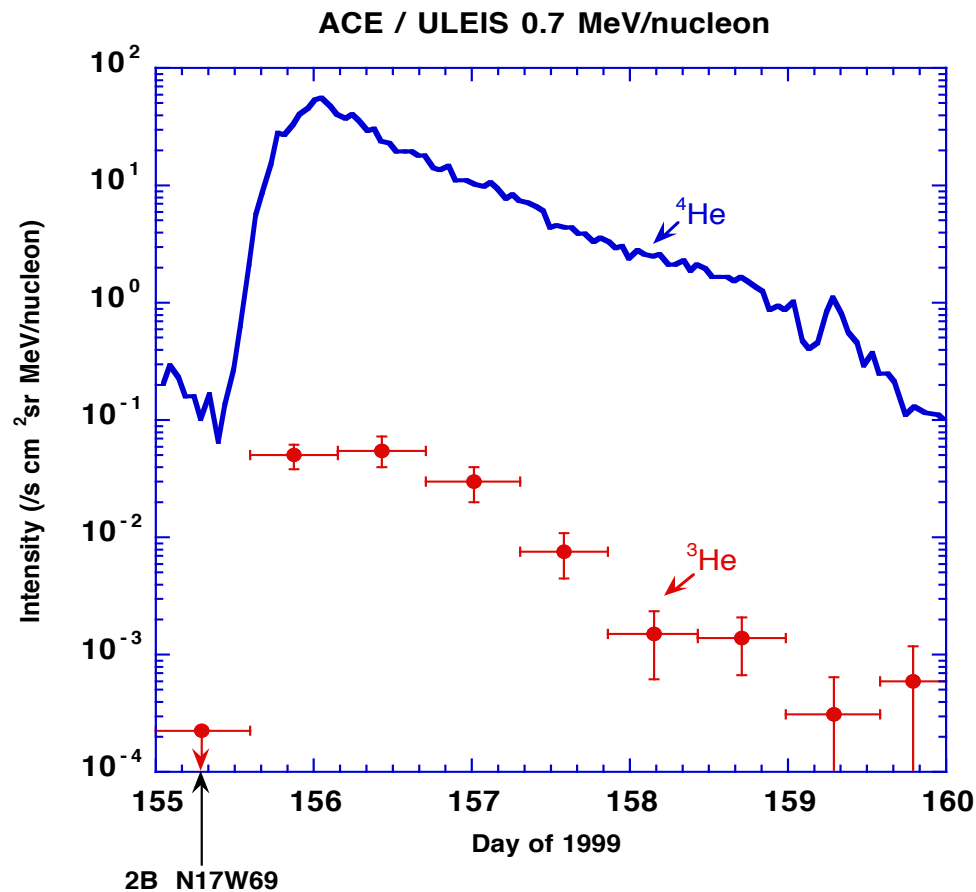
Heavy ion enhancements seen in solar energetic particles (Mason et al., 2004). Could be due to wave-particle interactions.



Mason et al., 2004

^3He in Gradual SEP Events

^3He abundance in ~50% of large SEP events \gg solar wind value of $\sim 5 \times 10^{-4}$



(Mason et al. 1999 ApJ, 525, L133).

^3He During Quiet-times

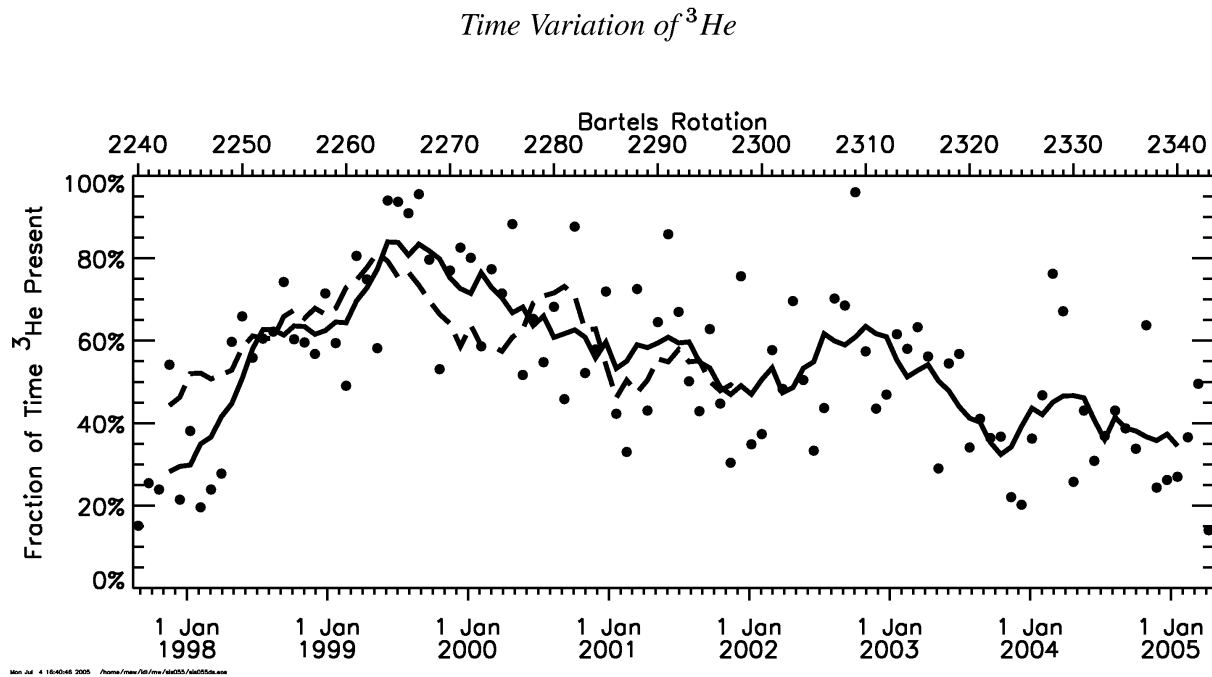


Figure 2. Time variation of the fraction of time when ^3He from impulsive SEP events is observed at 1 AU in any of the four energy intervals analyzed. The dashed curve is the smoothed result from our previous study [2] study. The solid curve is the smoothed result from the present study. The points indicate values obtained for individual Bartels rotations.

Widenbeck et al., ICRC, 2005

119

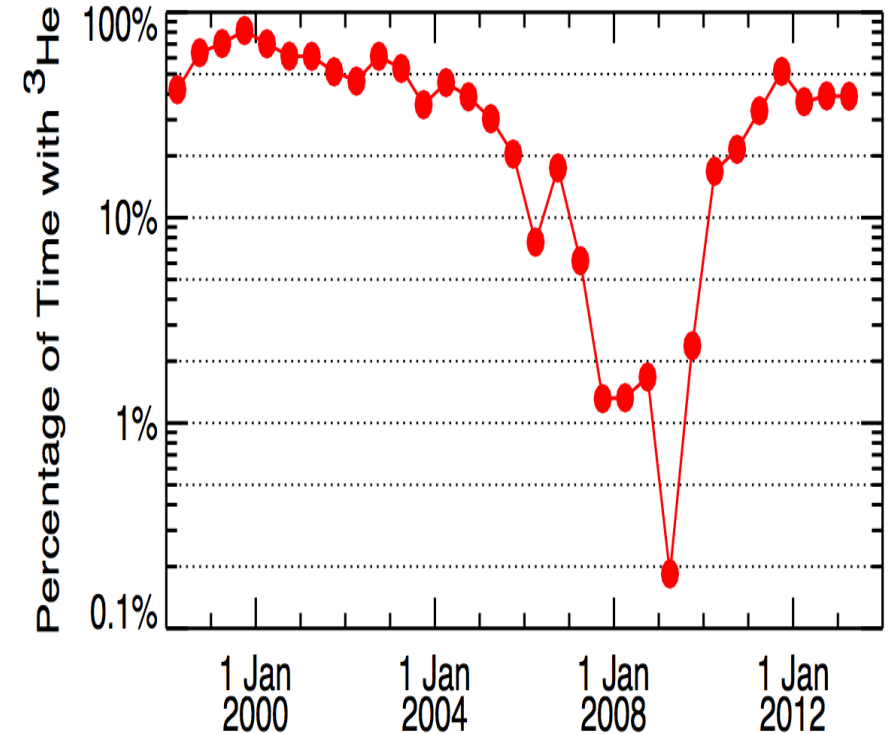
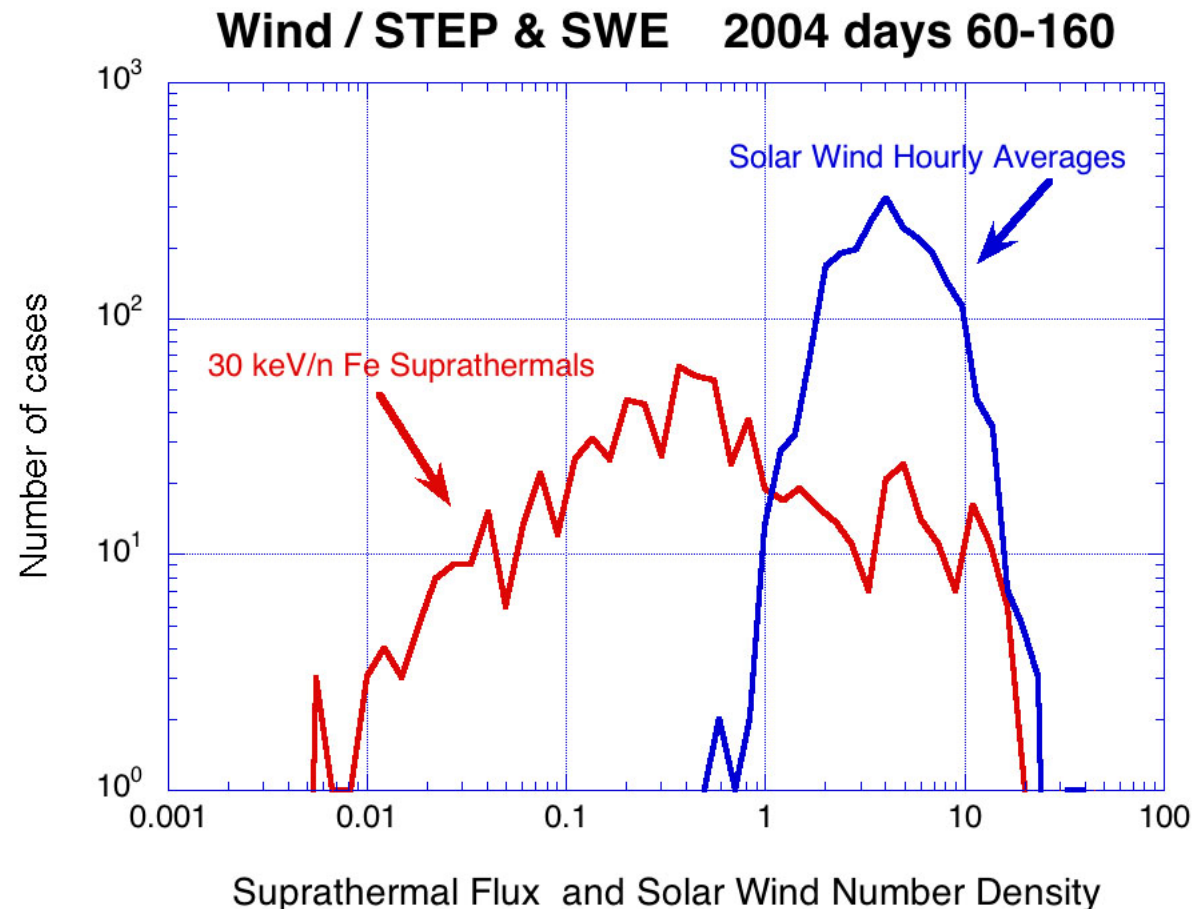


Figure 2. Average percentage of time that ^3He was detected at ACE during each 6-month period from the start of 1998 through mid-2013. All four of the energy intervals illustrated in Fig. 1 were used in identifying times when ^3He was present.

Widenbeck et al., AIP CP, 2014

The mystery of huge heavy ion (Fe) variations in intensity

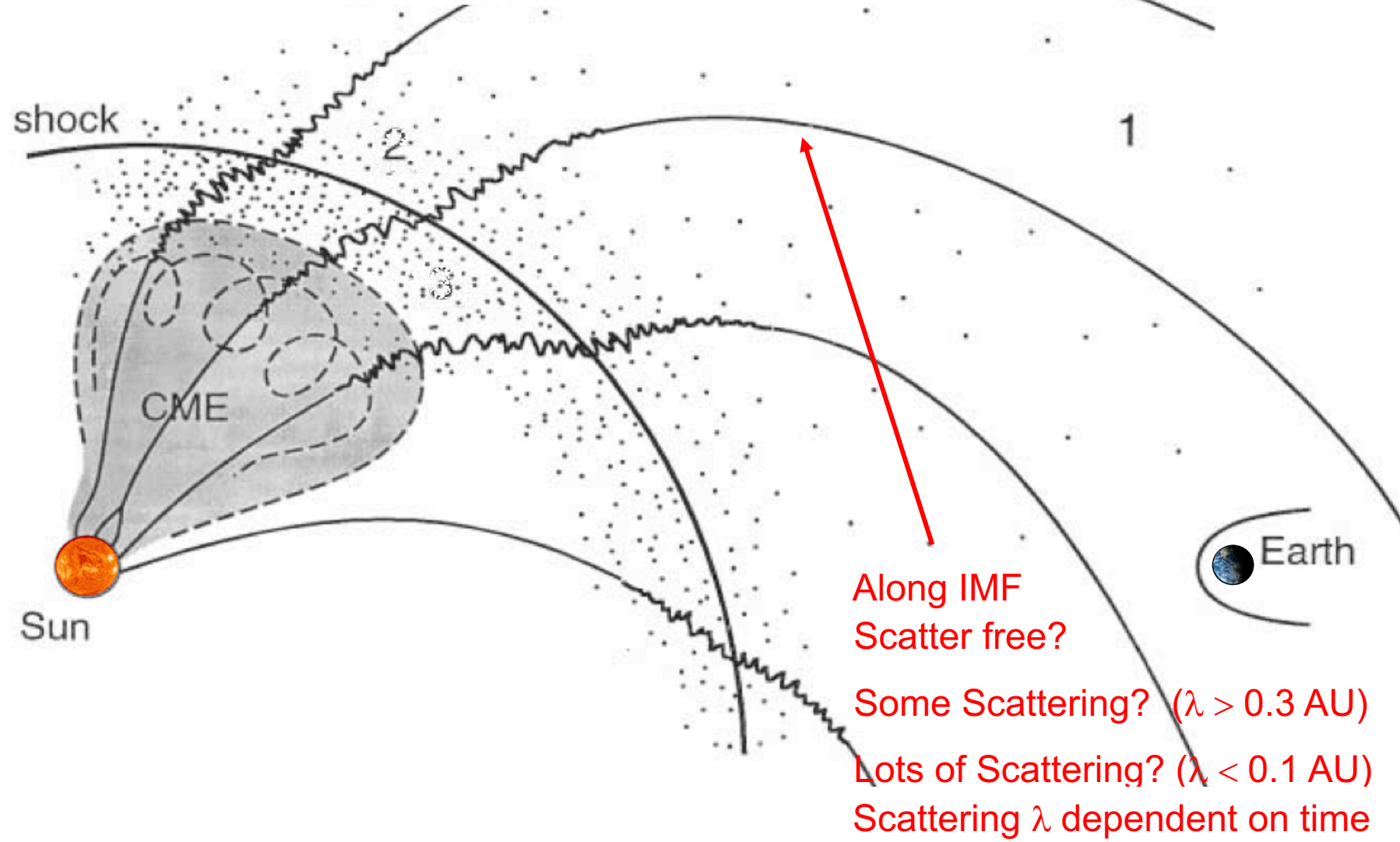


Peak intensities in shock events vary over a range of $\sim 10^4$

- not explained by CME speed*
- not explained by shock acceleration models*
- not explained by solar wind number density which does not change nearly as much*

Many researchers have suggested important role of suprathermal seed population

Transport to 1 AU



Along IMF
Scatter free?

Some Scattering? ($\lambda > 0.3$ AU)

Lots of Scattering? ($\lambda < 0.1$ AU)

Scattering λ dependent on time

And location?

Adapted from Marty Lee, *ApJS*, 158, 2005

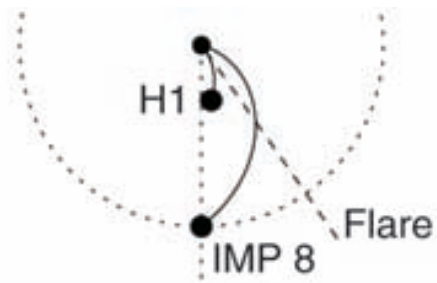
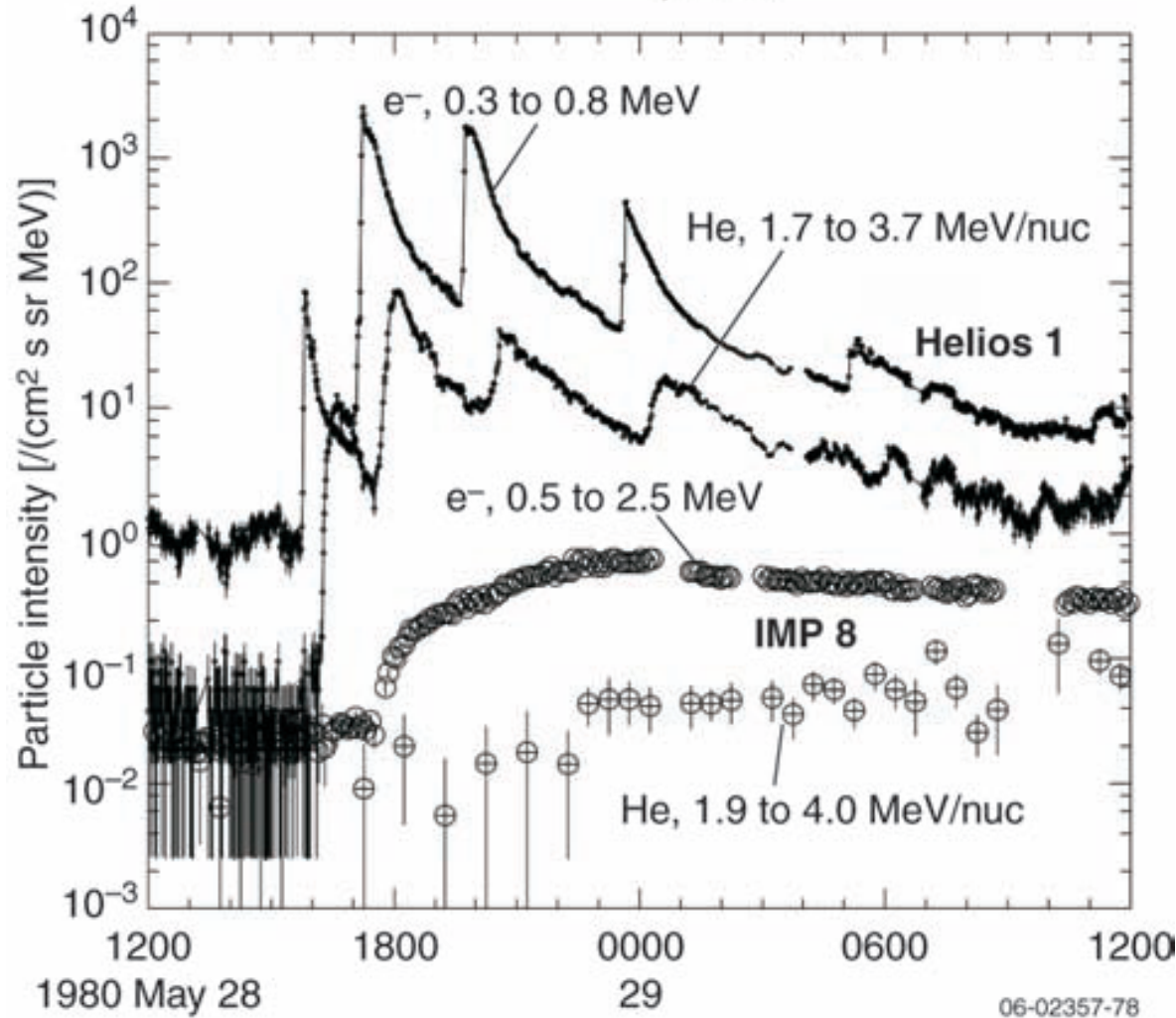


Illustration of the effect of transport on particles

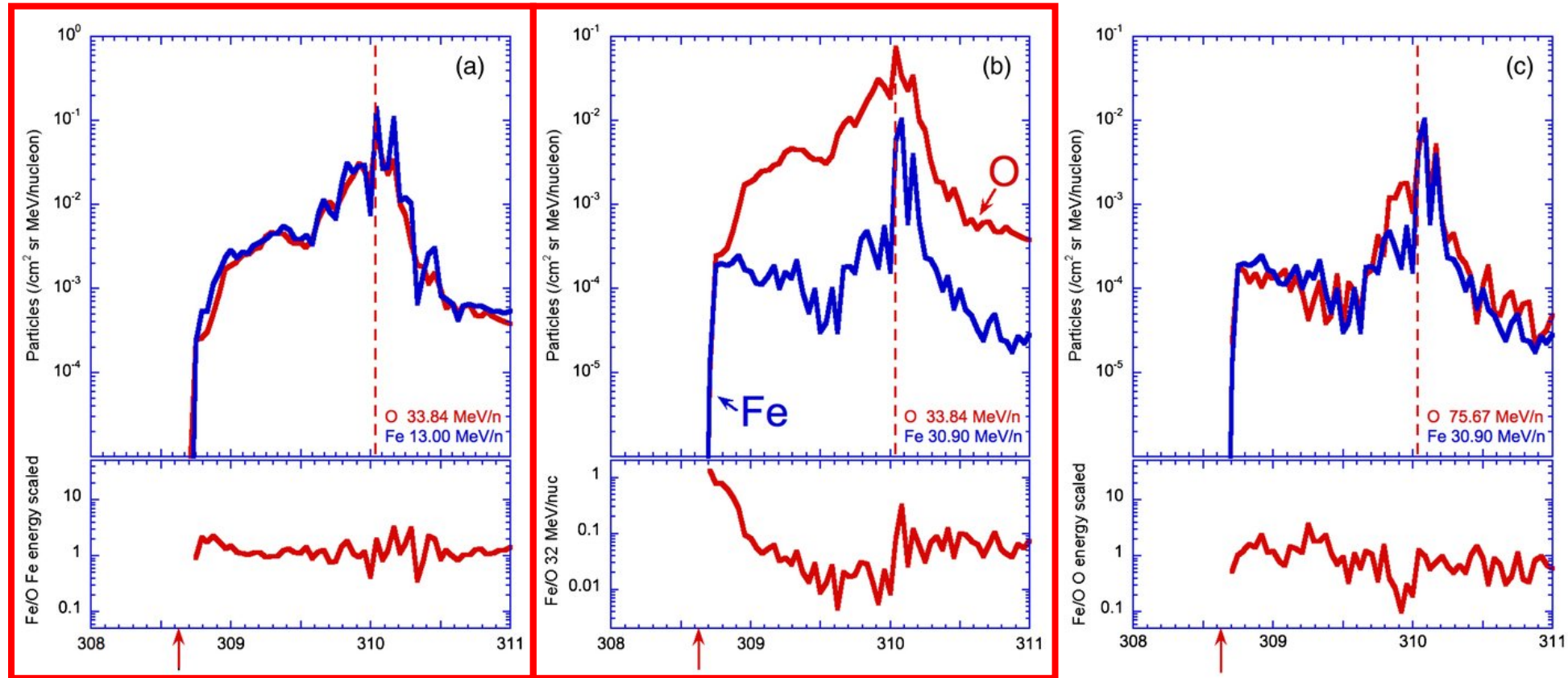


Need Solar Orbiter, PSP to disentangle transport effects from injection effects.

Wibberenz & Cane, 2006

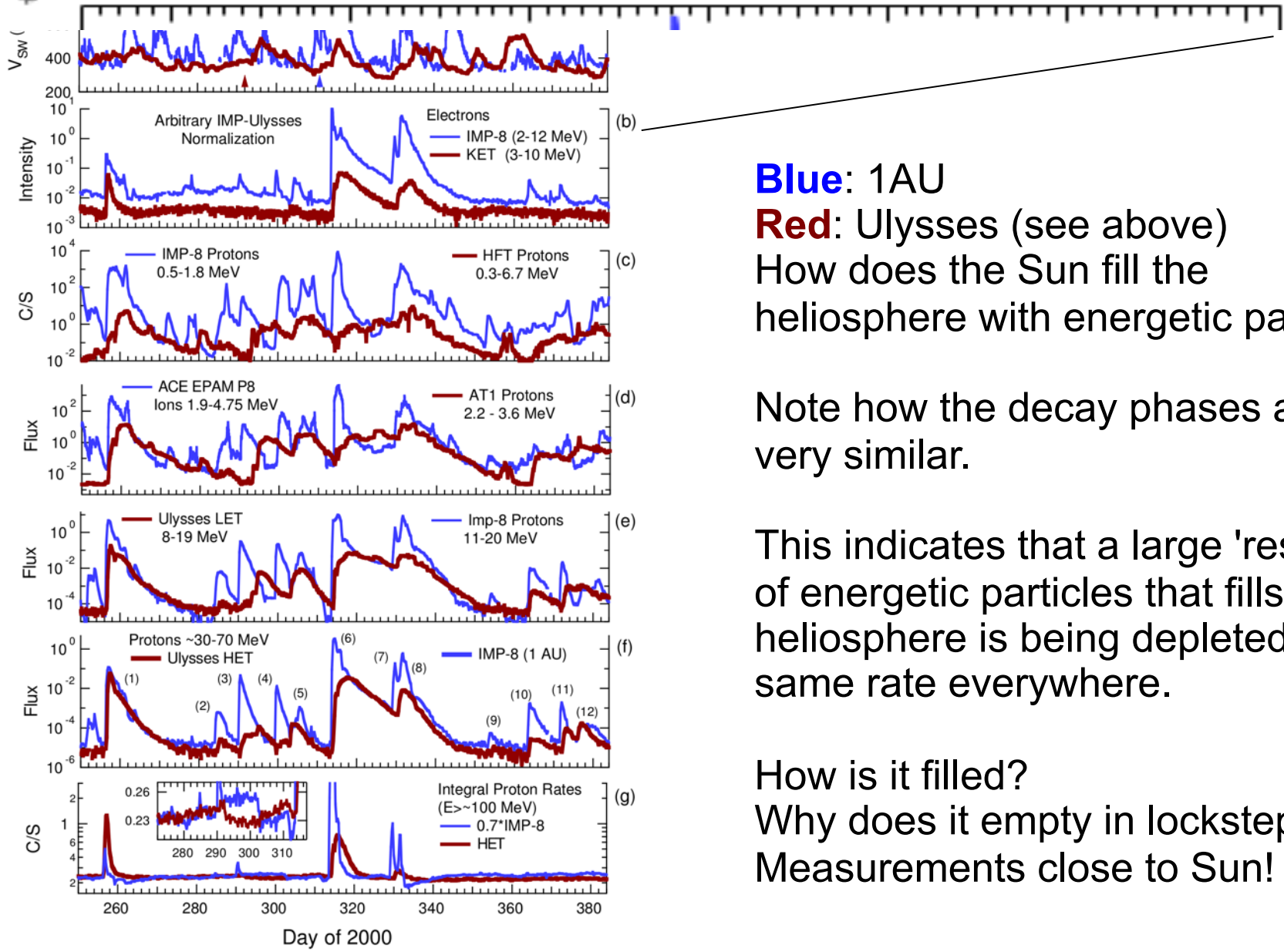
Transport effects

- Common energy organizes better than velocity
 - Maybe rigidity is better



Mason et al. 2012

R (AU)	2.77	2.64	2.50	2.36	2.22	2.08	1.94
Lat. (°)	-71.5	-74.7	-77.7	-79.9	-79.7	-76.5	-71.0
$\Delta\phi_p$ (°)	85.6	82.0	84.3	95.7	115.6	130.5	135.1



Blue: 1AU
Red: Ulysses (see above)
 How does the Sun fill the heliosphere with energetic particles?

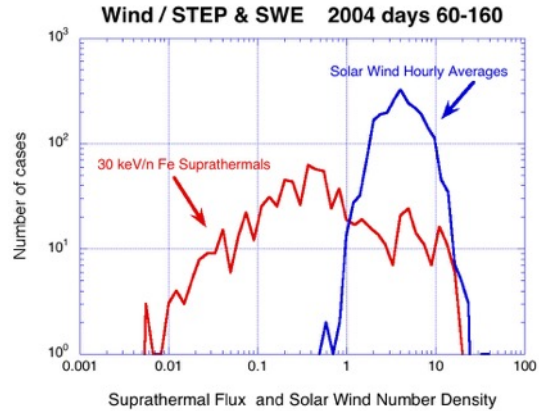
Note how the decay phases are all very similar.

This indicates that a large 'reservoir' of energetic particles that fills the heliosphere is being depleted at the same rate everywhere.

How is it filled?
 Why does it empty in lockstep?
 Measurements close to Sun!

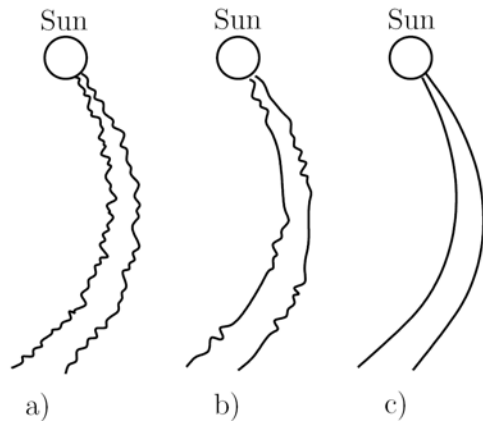
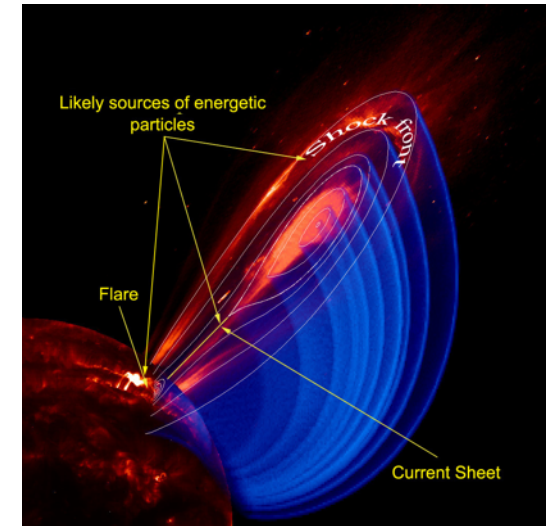
McKibben et al., 2003

Outstanding Question

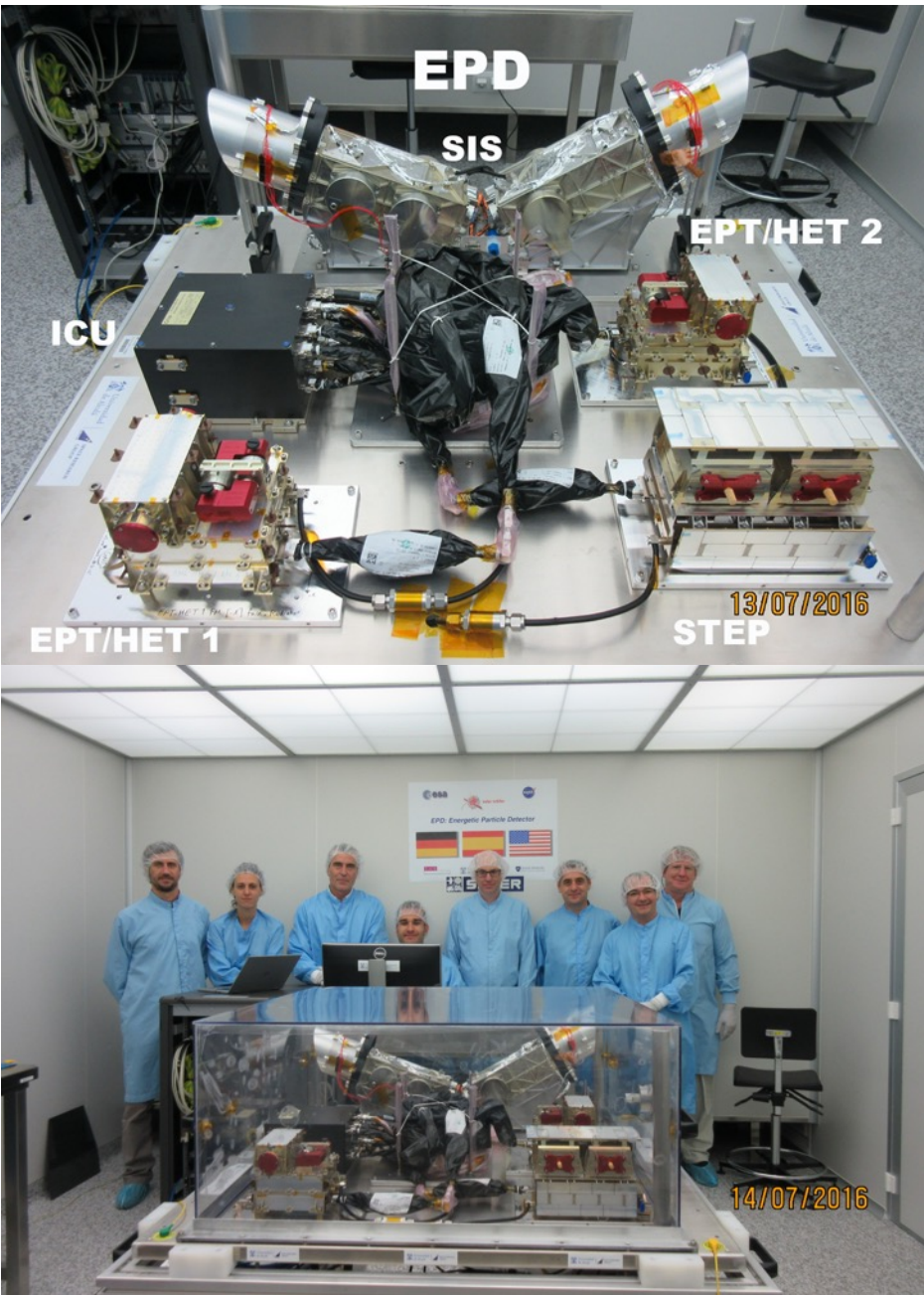


Injection (seed population important)

Acceleration (various processes)

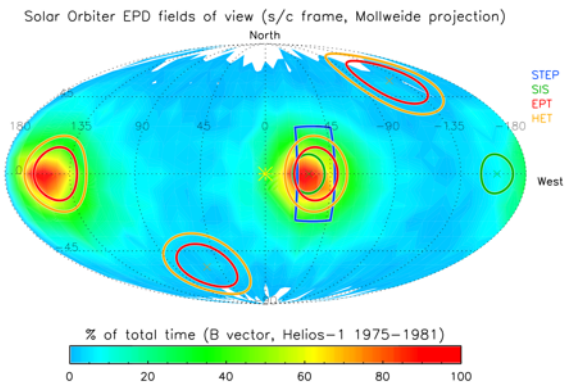
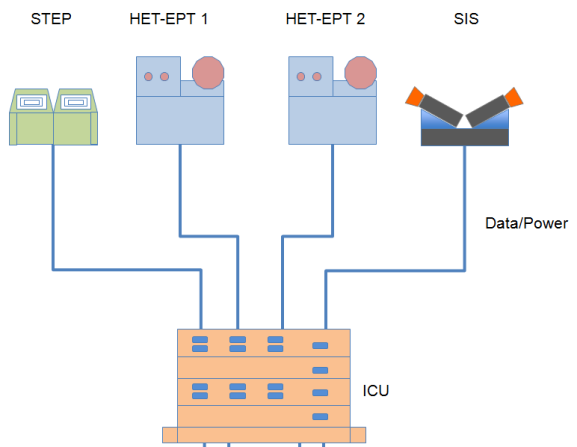
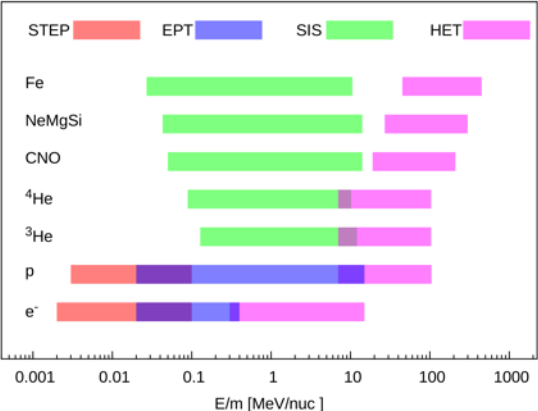


Transport (which smears out signatures)



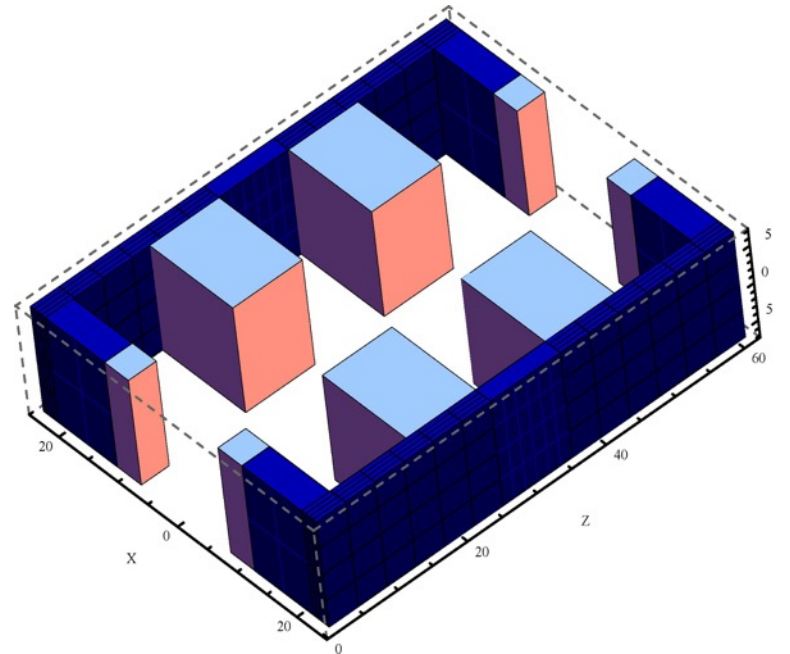
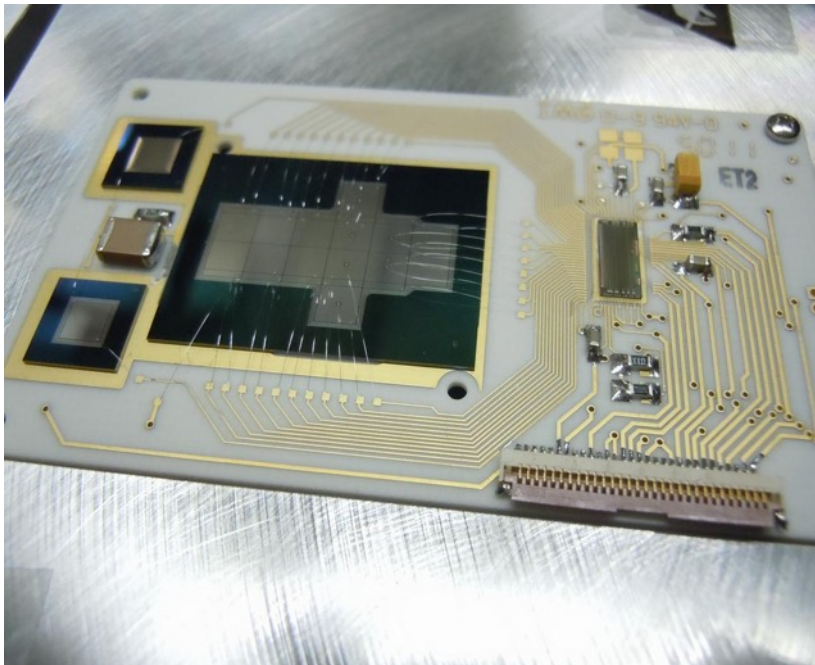
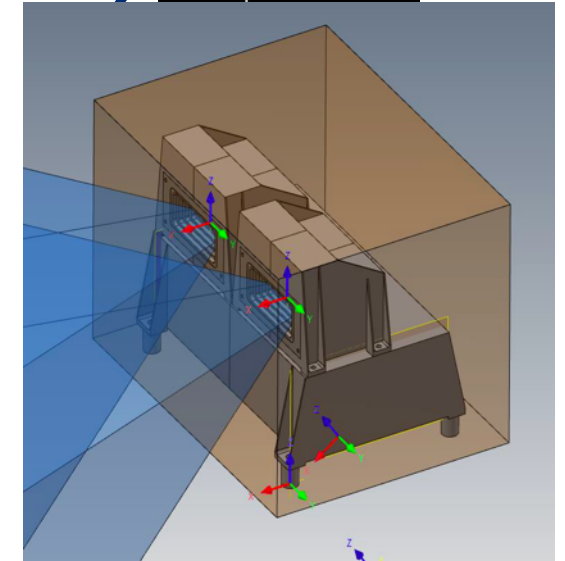
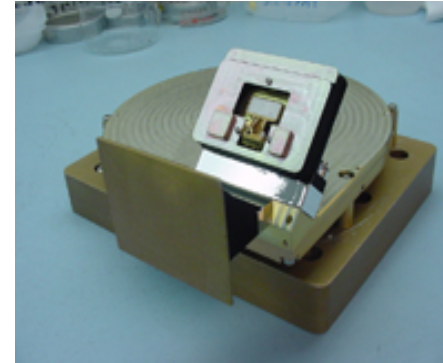
EPD is a suite of four sensors that measures suprathermal to high-energy particles:

- STEP (Suprathermal Electrons and Proton)
- EPT (Electron-Proton Telescope)
- SIS (Suprathermal Ion Spectrograph)
- HET (High-Energy Telescope)

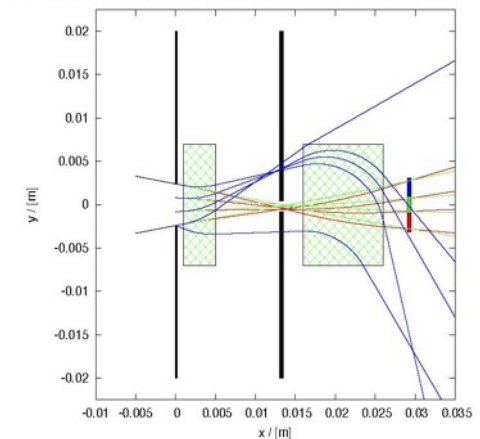


SupraThermal Electron and Proton (STEP)

- S/C mounted
 - STEP will measure e^- [0.002-0.1MeV] and p^+ [0.003-0.1MeV]
- Instrument Heritage:
 - STEP has heritage of STEREO/STE



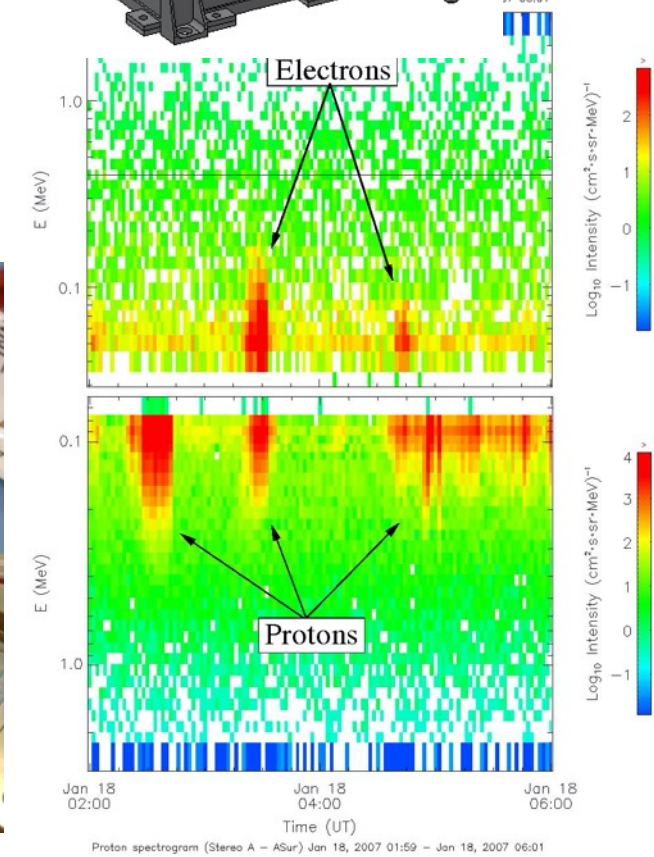
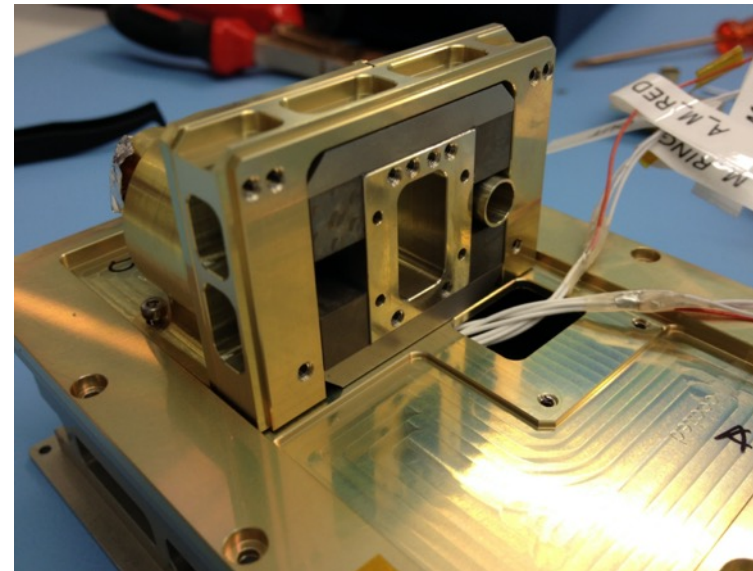
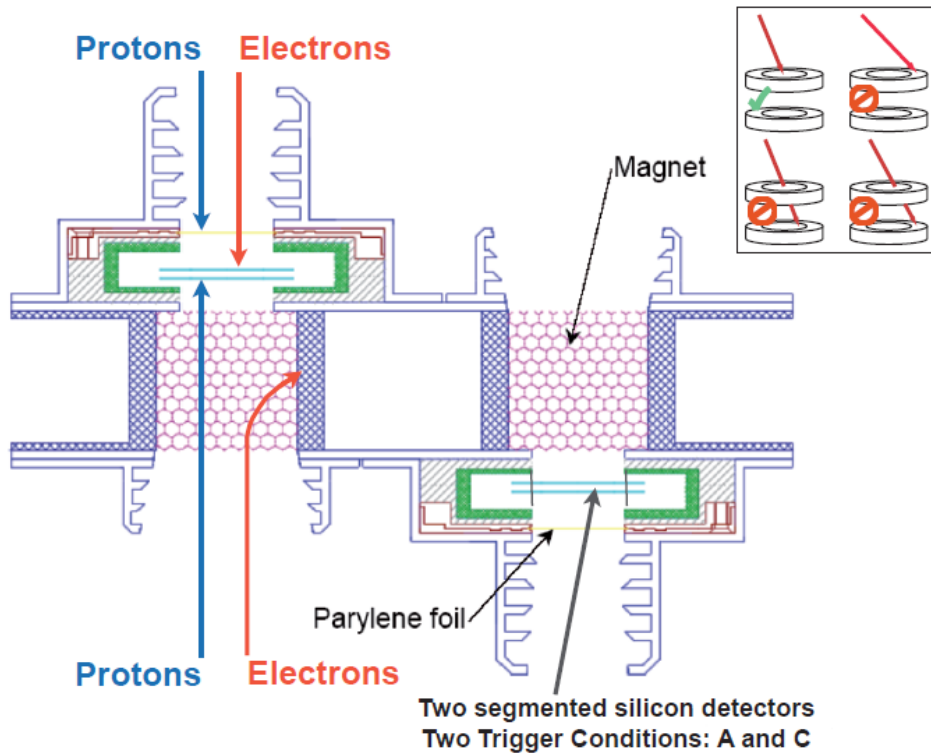
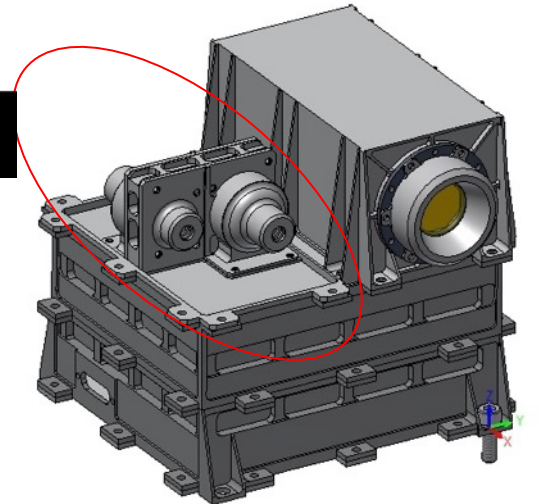
particle trajectories in $B=0.1T$, protons at 4000eV in red, at 100keV in green, electrons at 100keV in blue



Electron Proton Telescope (EPT)



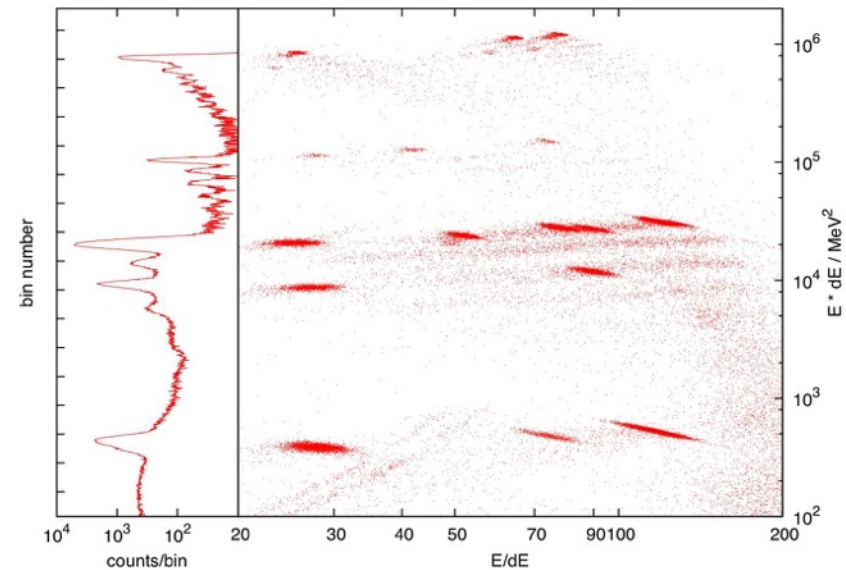
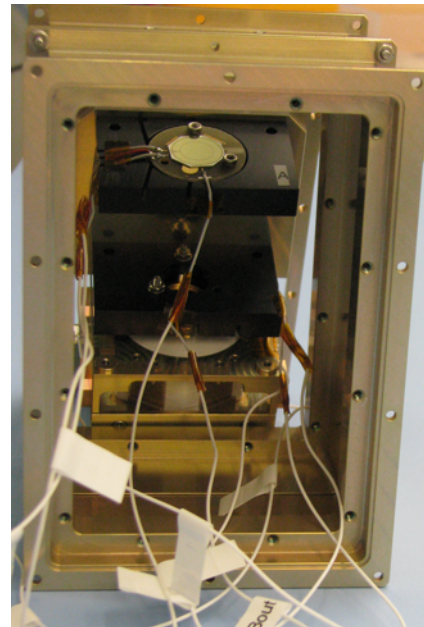
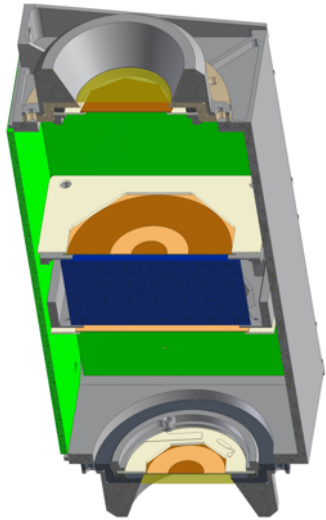
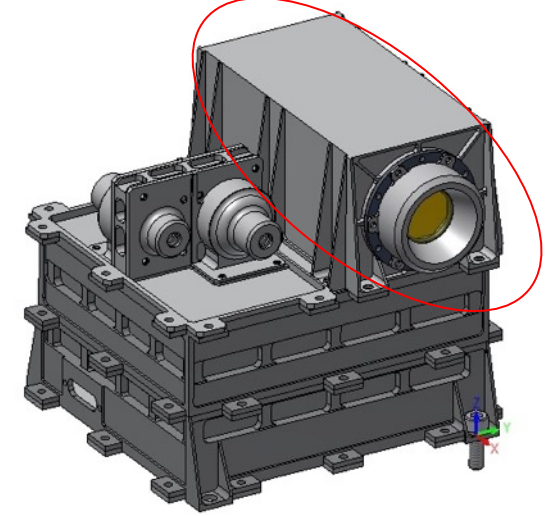
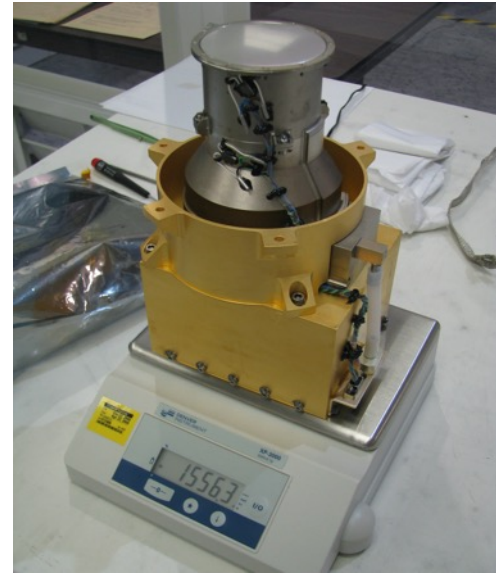
- Measures electrons and protons
 - EPT will measure e^- [0.02-0.7 MeV], p^+ [0.02-7 MeV]
 - Four view directions (in and out of orbital plane) with two units
 - EPT and HET sensors share the same Electronic Box
- Instrument Heritage:
 - EPT has direct heritage from STEREO/SEPT



High-Energy Telescope (HET)

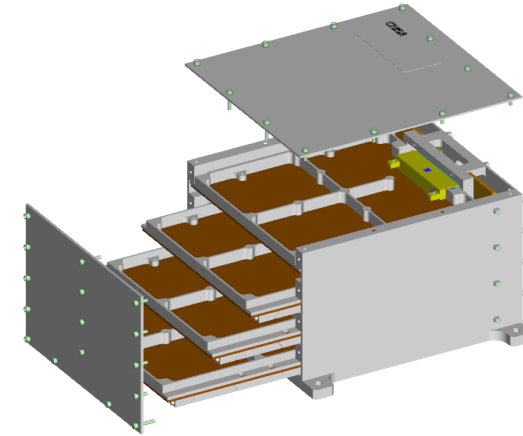
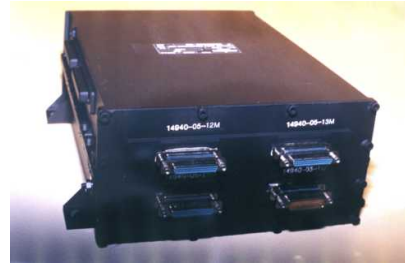


- Measures e^- [0.3-20MeV], p^+ [10-100MeV] and ions [50-200MeV/nuc **Z-dependent]
 - Two view directions (in and out of orbital plane) with two units
 - EPT and HET sensors share the same Electronics Box
- Instrument Heritage:
 - HET has heritage from the MSL/RAD instrument



Integrated Control Unit (ICU)

- Provides a single point of connection between the S/C and all the EPD sensors
 - Power I/F
 - Data I/F
 - Cold redundant configuration
- Heritage
 - ICU has heritage from SOHO/COSTEP CDPU



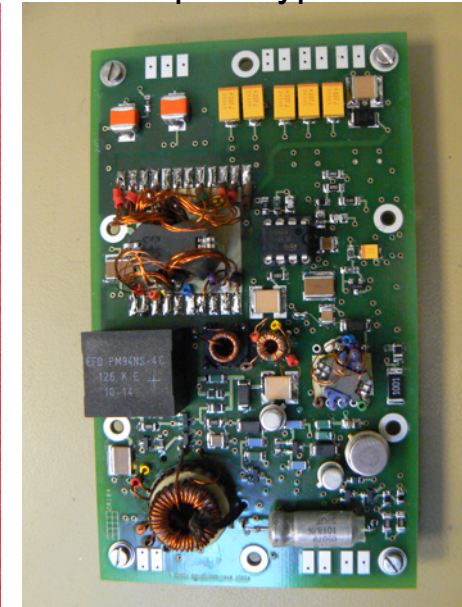
ICU emulator



ICU prototype

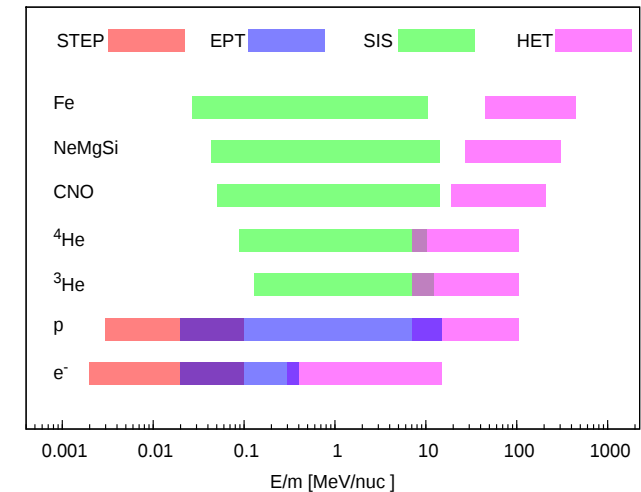
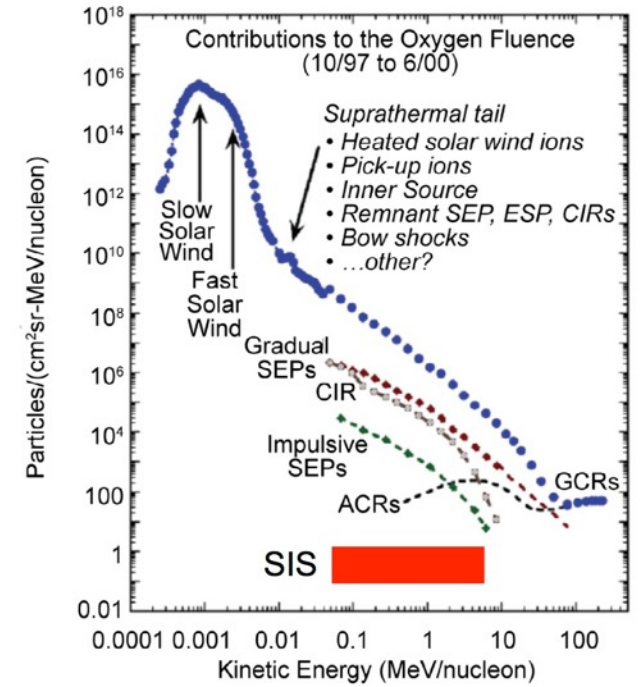
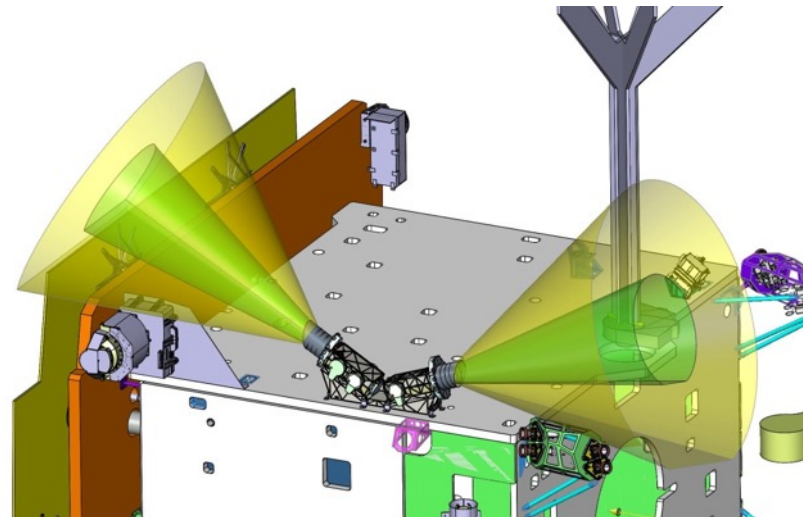
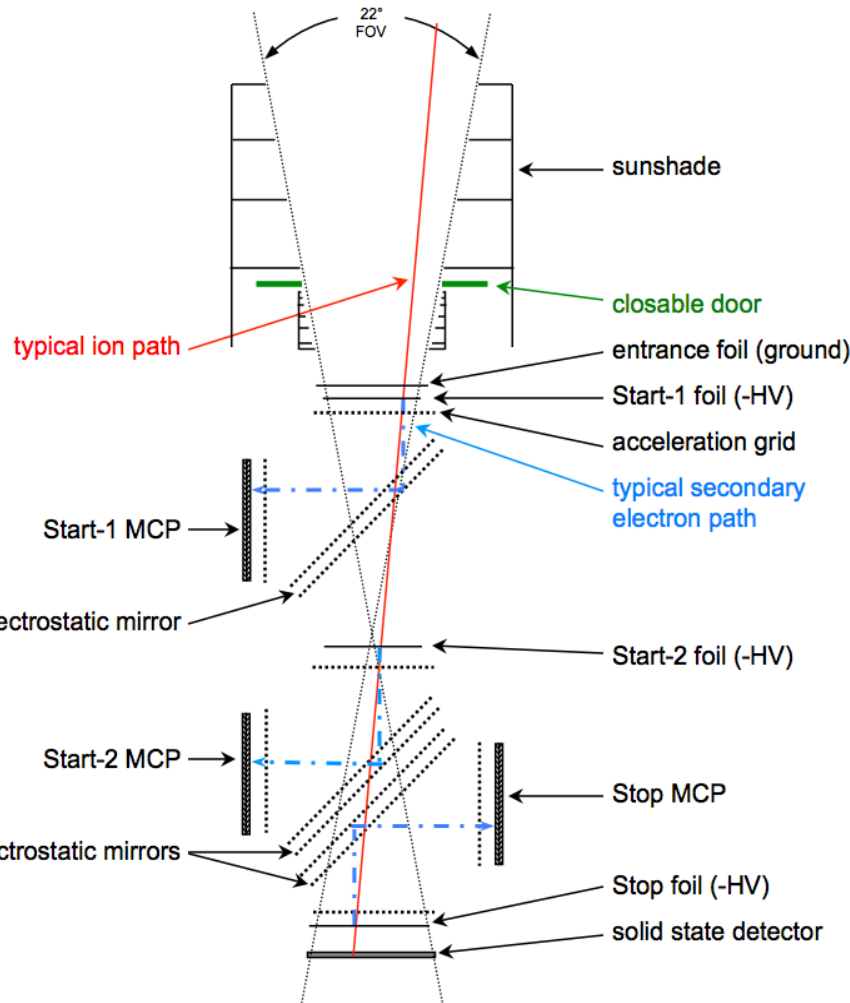


LVPS prototype

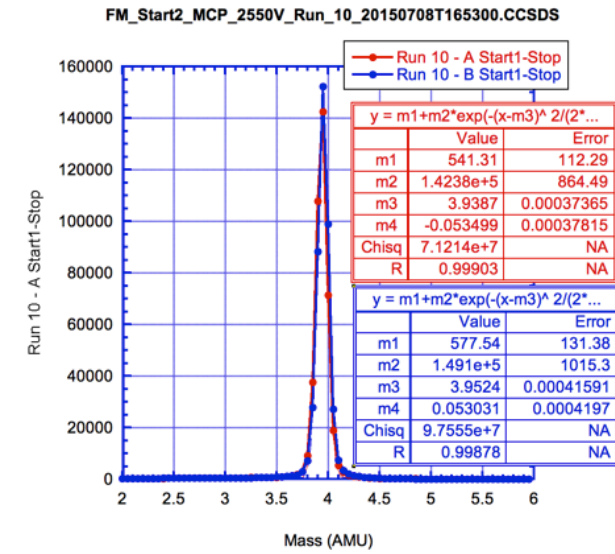
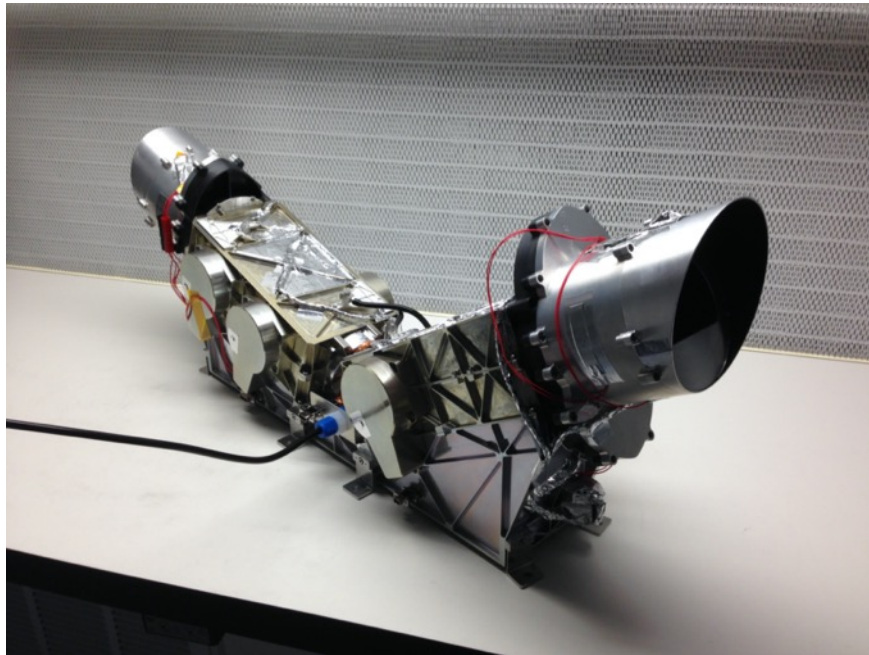


Suprathermal Ion Spectrograph (SIS)

- SIS is a Time-of-Flight by Energy high-resolution ion mass spectrometer
- Two identical telescopes measure the ion mass and energy
- Same measurement technique as was used on ACE/ULEIS



SIS-Flight Model



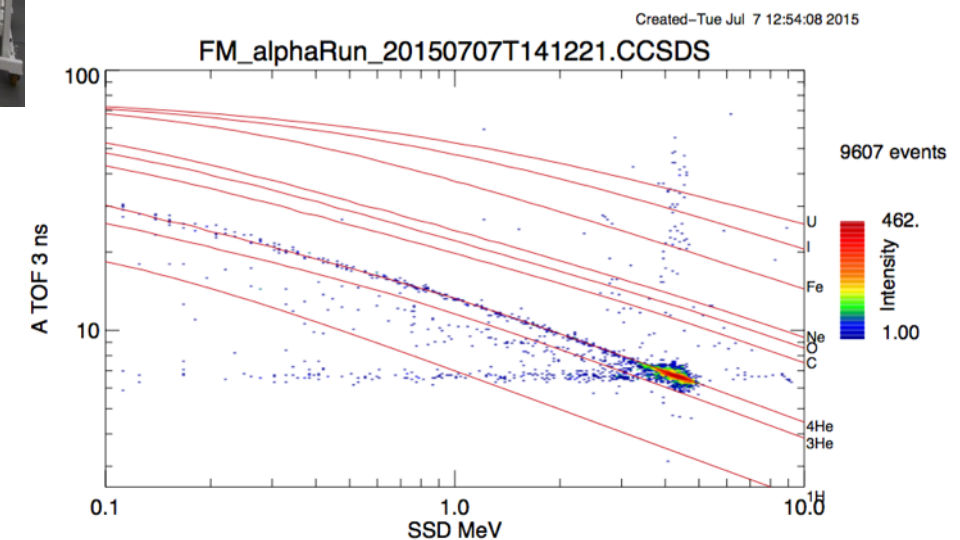
He histograms with TOF-1

m/sigma-m:

A telescope: 73.6

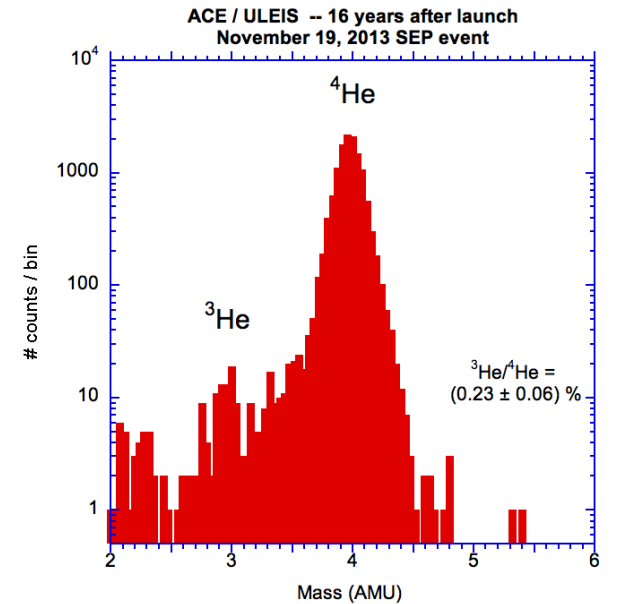
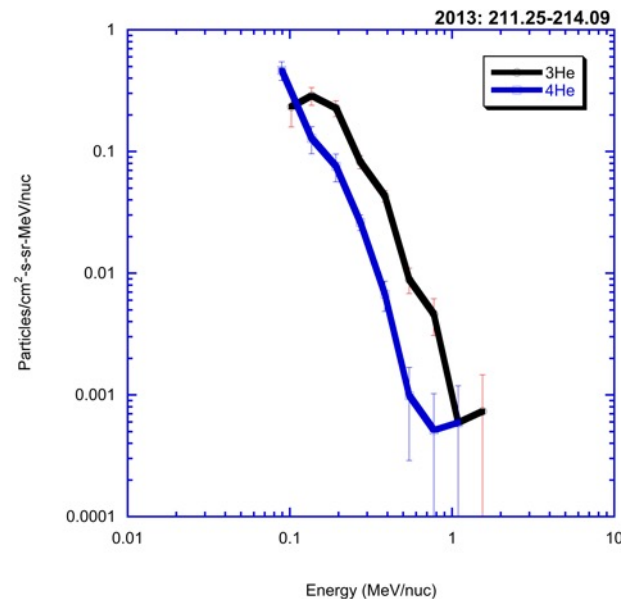
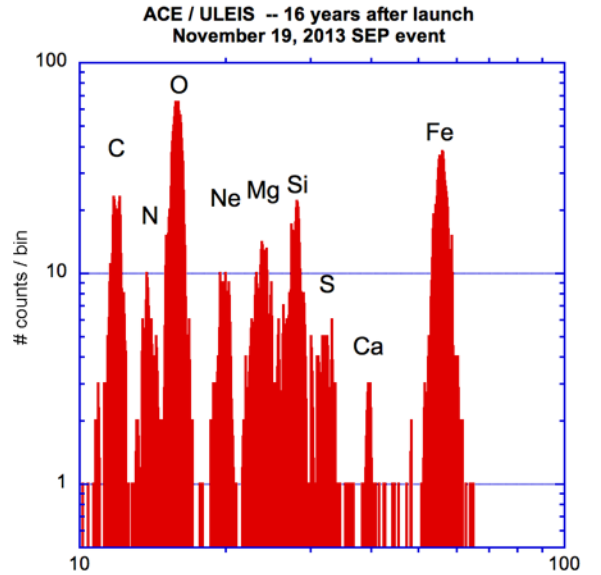
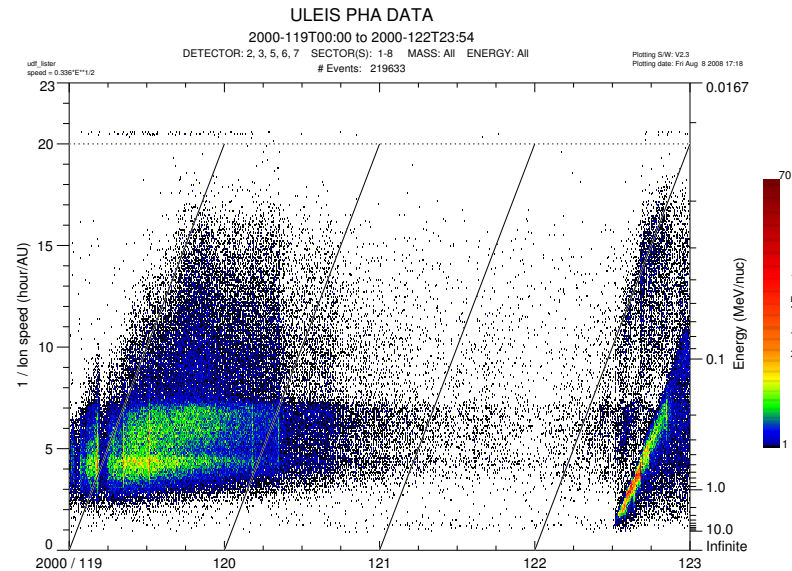
B telescope: 74.5

- SIS Flight Model
- FM met and exceeded all the L1 requirements:
 - Energy resolution (<5%)
 - Energy range (0.05 – 10 MeV/nuc)
 - Mass resolution ($m/dm > 50$)
 - Etc.

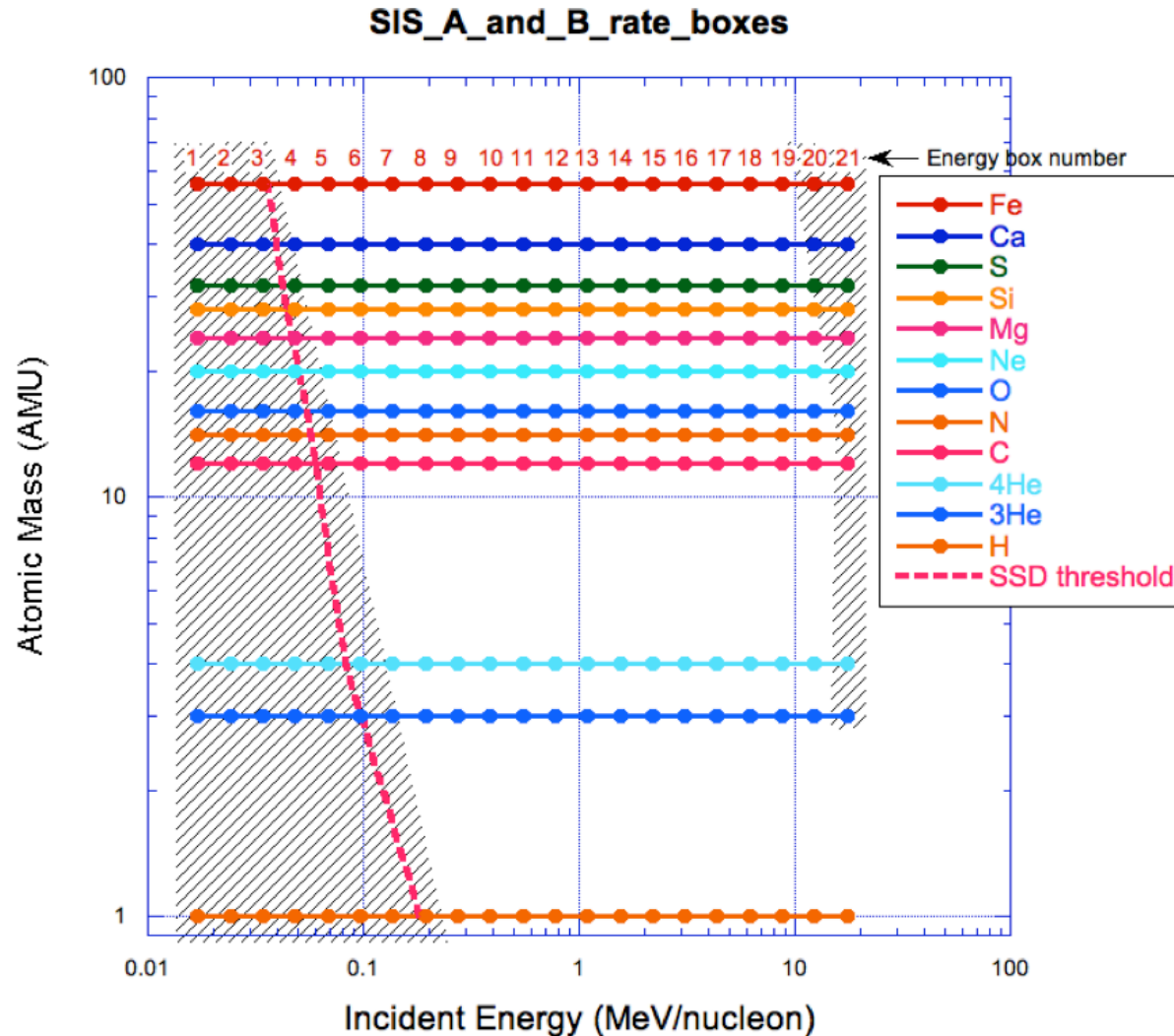


SIS Data Products

- **Energy Range**
 - 50 keV/nuc – 10 MeV/nuc
- **Three science data products**
 - Matrix rates for 12 major elements (H, 3He, 4He, C, N, etc)
 - Pulsed Height Analyzer Events
 - Helium Histogram ***New***
- **Time Cadences**
 - Nominal 30 s
 - Burst 3 s
 - Quick look 30 minutes
- **Two look directions ***New*****
 - Sunward 30° off the deck; Anti-sunward 20° off the deck
 - Each has an iris mechanism to control aperture from 100% to 1%

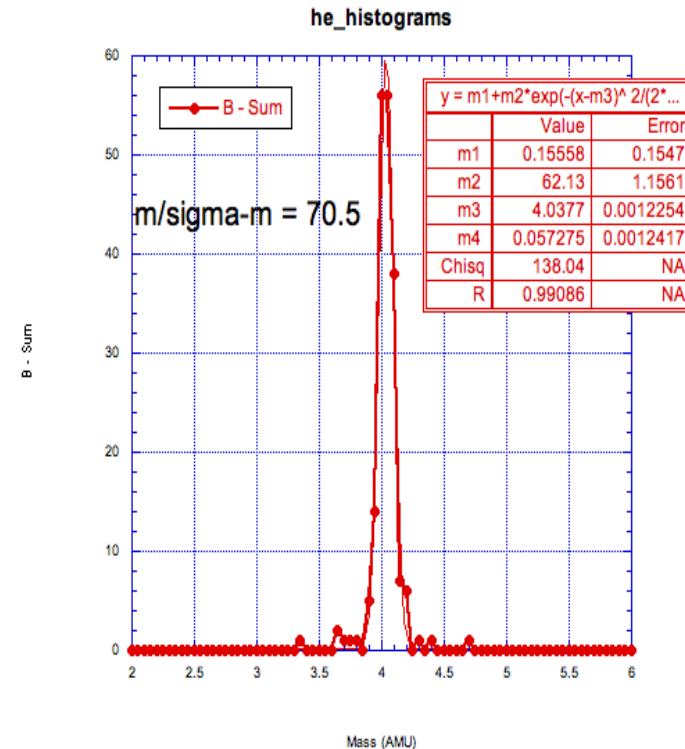
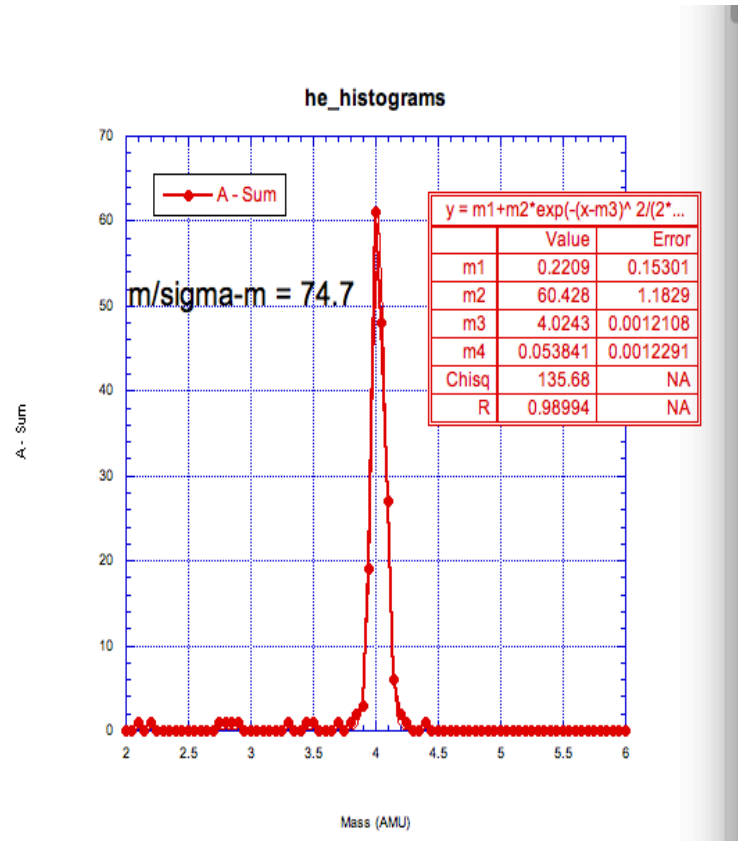


SIS Matrix Rate Box



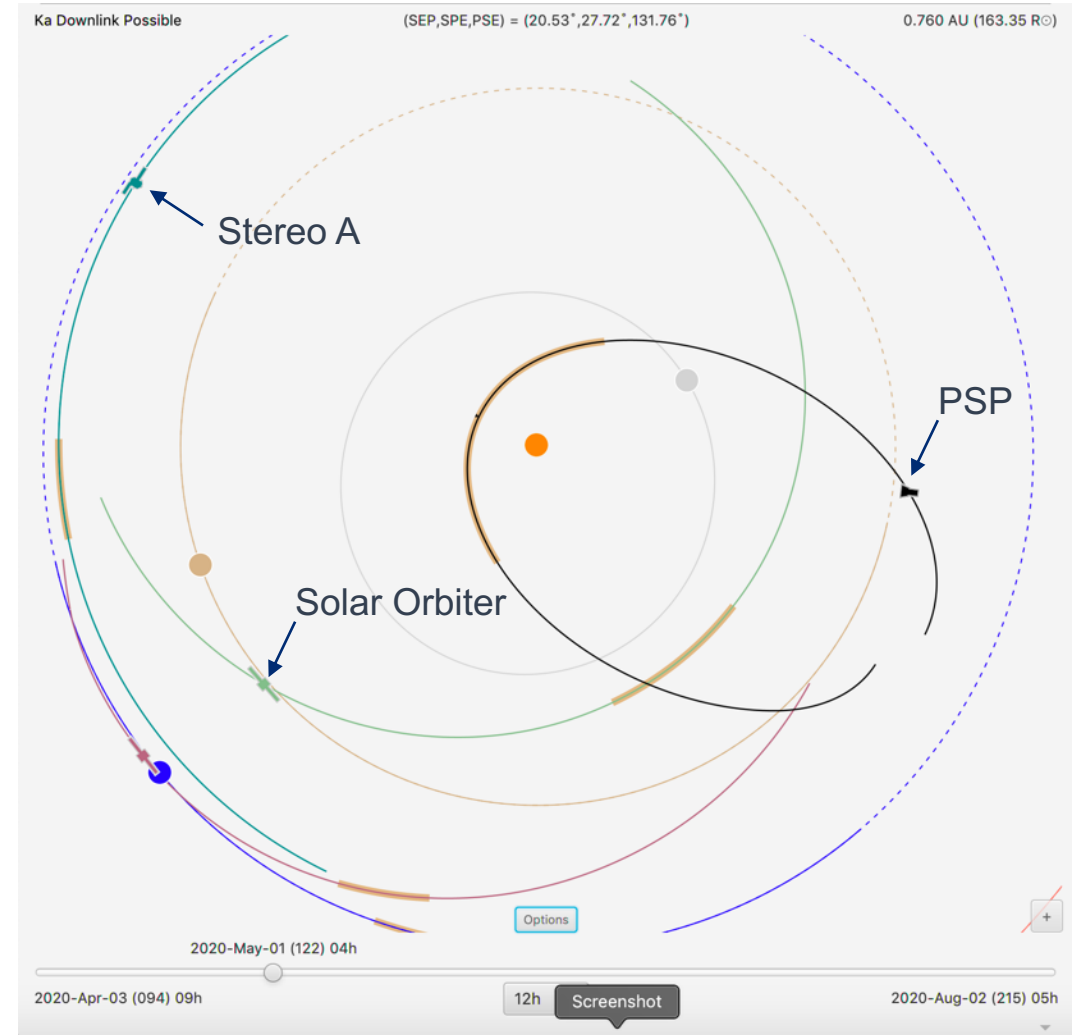
Energy box #	Elo Incident energy (MeV/nucleon)	Ehi Incident energy (MeV/nucleon)	Eaverage (MeV/nucleon)
0	0.0141	0.0200	0.0171
1	0.0200	0.0283	0.0241
2	0.0283	0.0400	0.0341
3	0.0400	0.0566	0.0613
4	0.0566	0.0800	0.0683
5	0.0800	0.1131	0.0966
6	0.1131	0.1600	0.1366
7	0.1600	0.2263	0.1931
8	0.2263	0.3200	0.2731
9	0.3200	0.4525	0.3863
10	0.4525	0.6400	0.5463
11	0.6400	0.9051	0.7725
12	0.9051	1.2800	1.0925
13	1.2800	1.8102	1.5451
14	1.8102	2.5600	2.1851
15	2.5600	3.6204	3.0902
16	3.6204	5.1200	4.3702
17	5.1200	7.2408	6.1804
18	7.2408	10.2400	8.7404
19	10.2400	14.4815	12.3608
20	14.4815	20.4800	17.4807

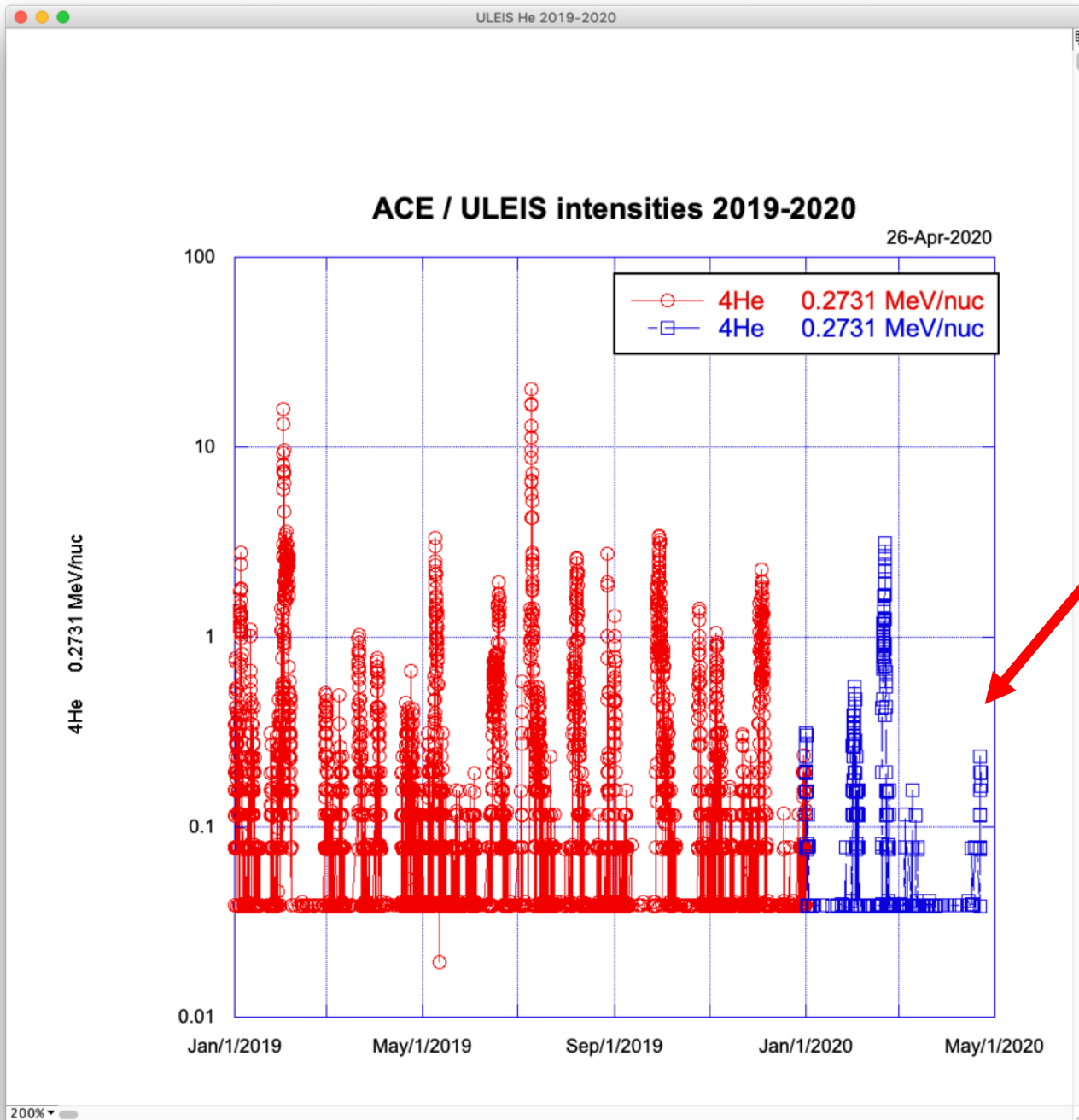
Helium Mass Histogram



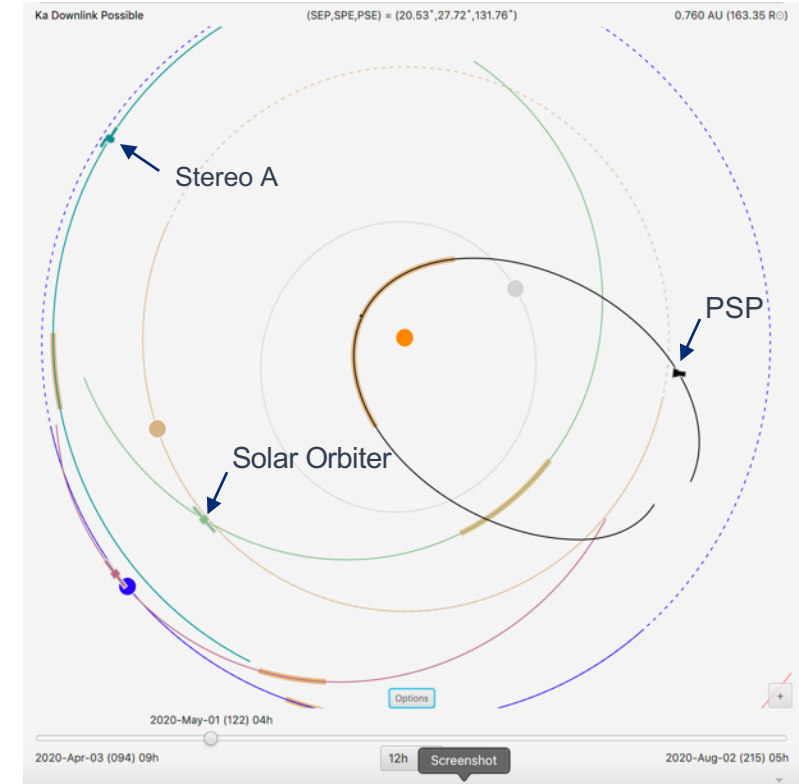
- Since the PHA telemetry is limited, this limits the timing resolution for $^3\text{He}/^4\text{He}$ ratio measurements.
- To help avoid this, He mass histograms are formed in the instrument allowing analysis of all the PHA events, not just the telemetered ones.
- The He mass histogram is:
 - mass range 2.0 - 6.0 AMU;
 - energy range 0.5 - 2.0 MeV/nuc

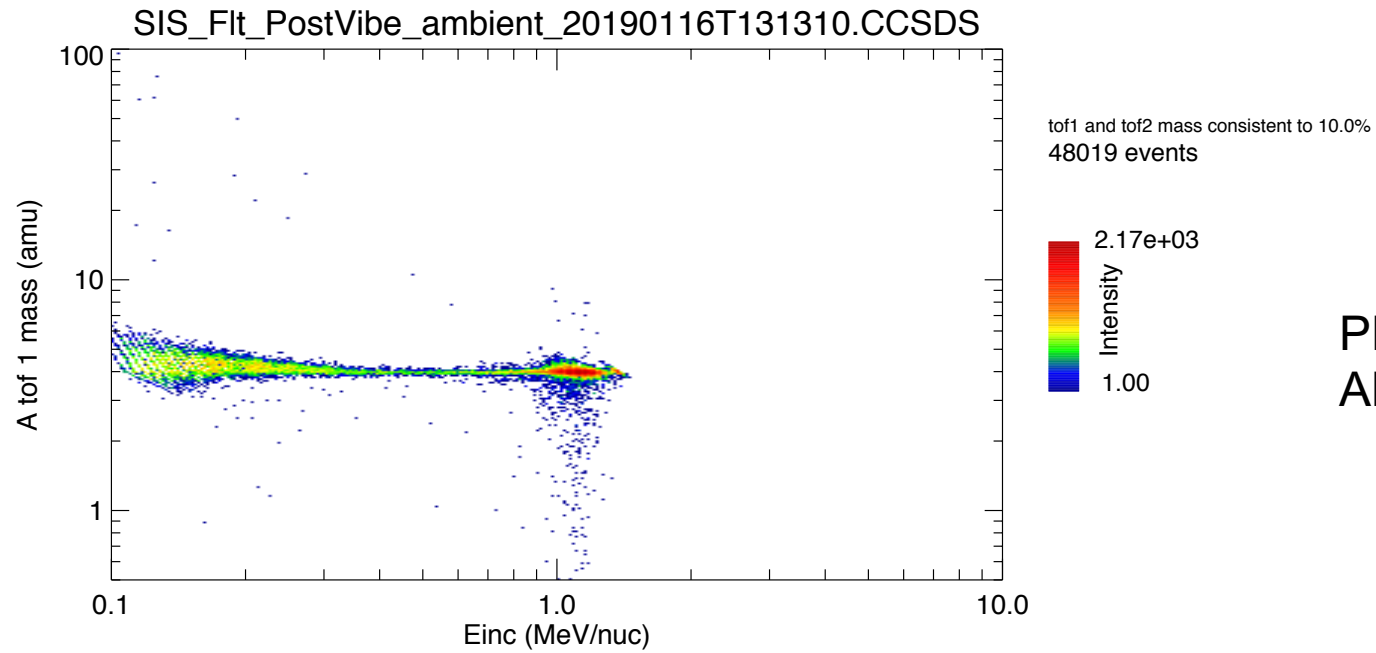
Launch: February 9 (10), 2020



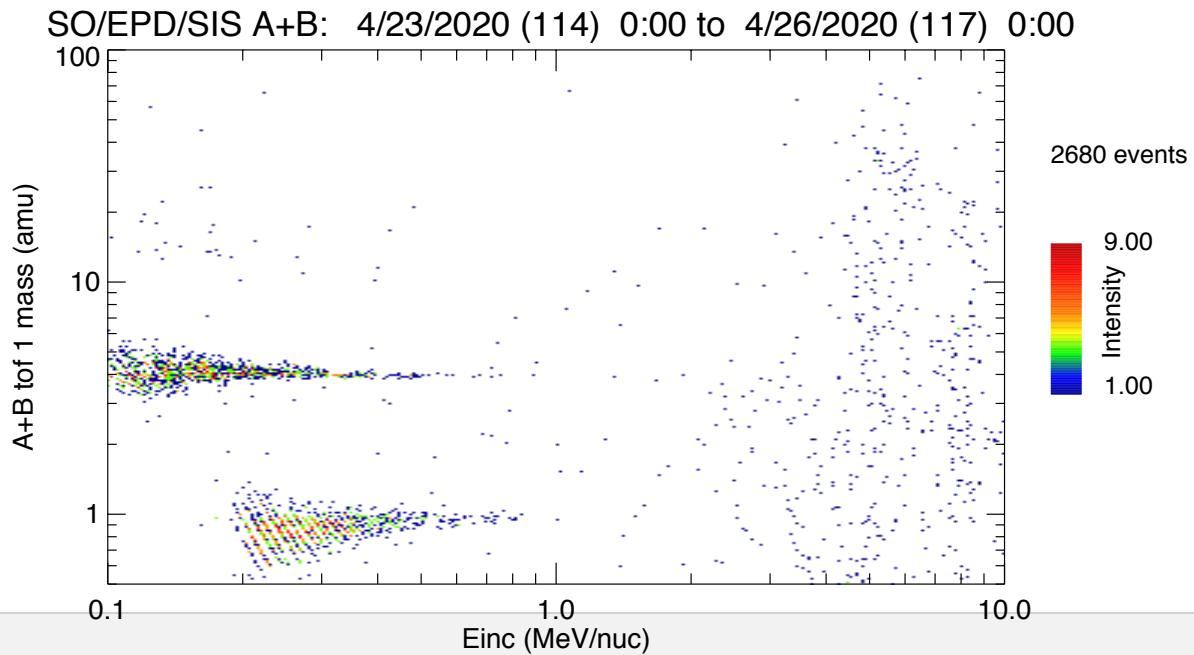


Energetic particle
Intensities very
Low right now

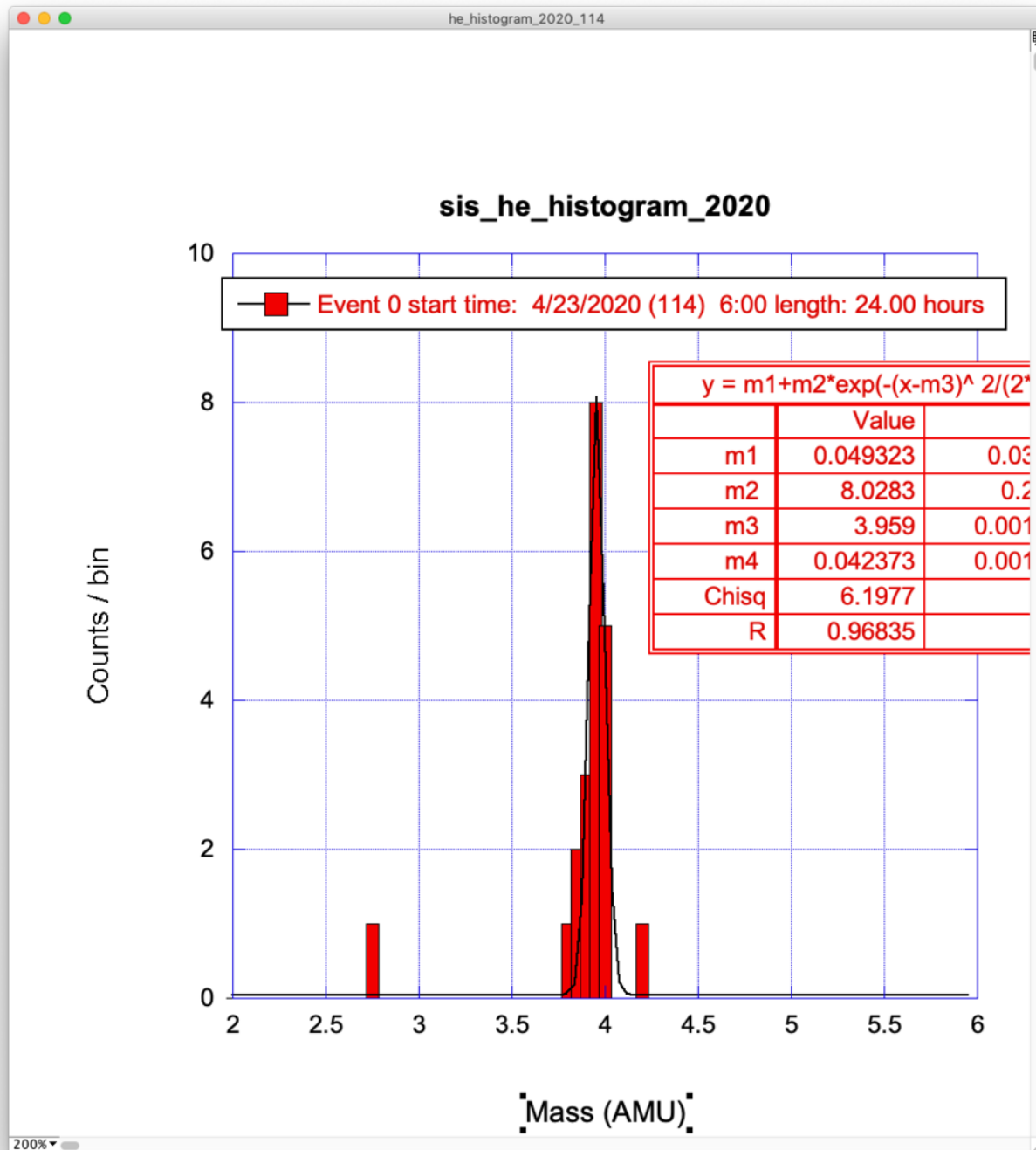




PRELAUNCH
ALPHA RUN



POST Launch
3 day sum after
HV ramp up

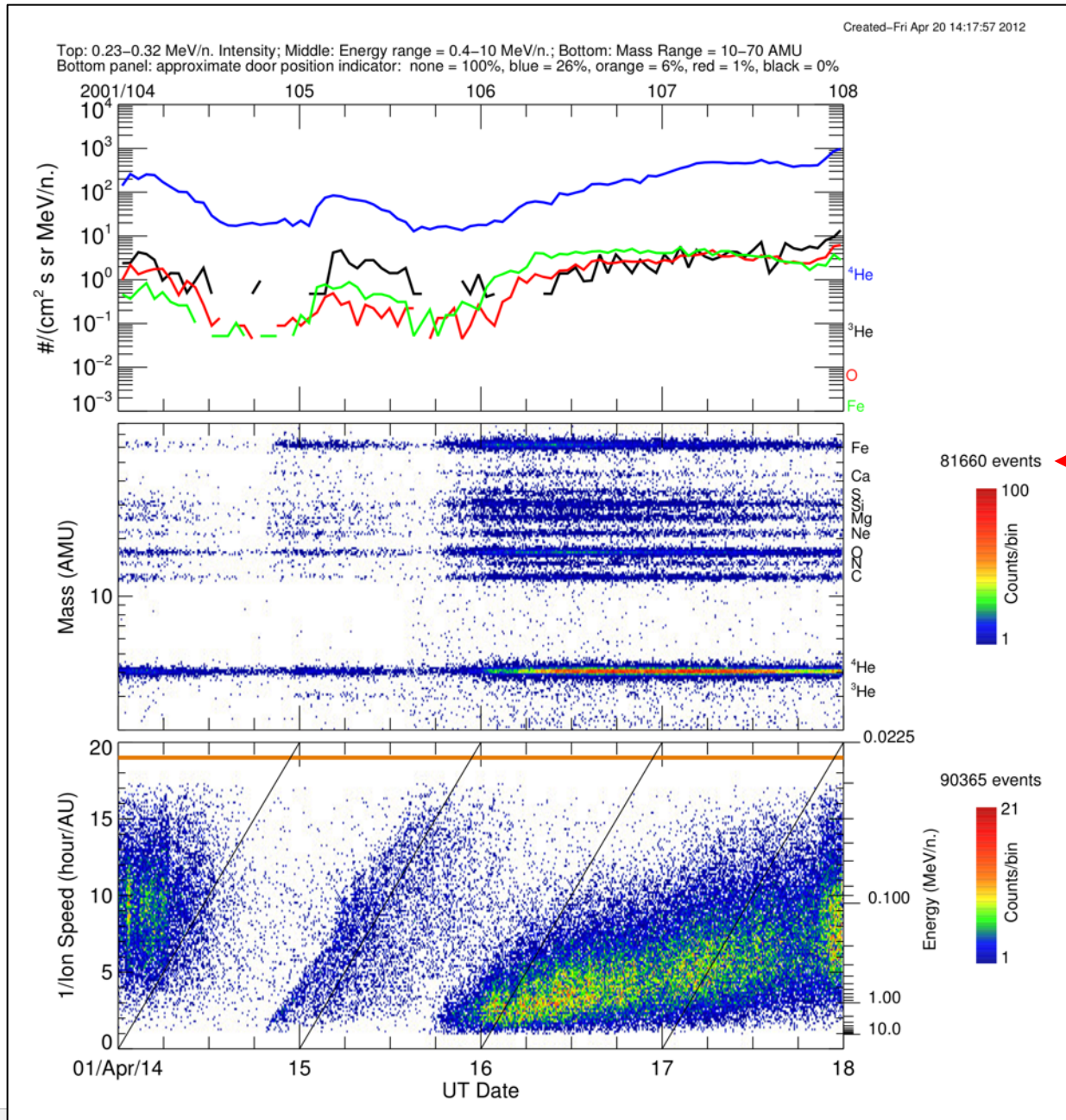


SIS He histogram

*24 hour sum over
Weak CIR*

*Most ions are near
0.5 MeV/nucleon*

*$m/\sigma - m =$
 $3.959/0.0424 =$
 93.4 consistent
With prelaunch
Values and science
Requirement
(poor statistics)*



Level 3 products are
Higher level data, such
As plots

These will be developed
by the EPD team in the
Future

Example of plot from
ACE/ULEIS posted at
ACE Science Center –
4 day plots for entire
Mission

Shows impulsive, CME
Related events during
Period of high solar
Activity

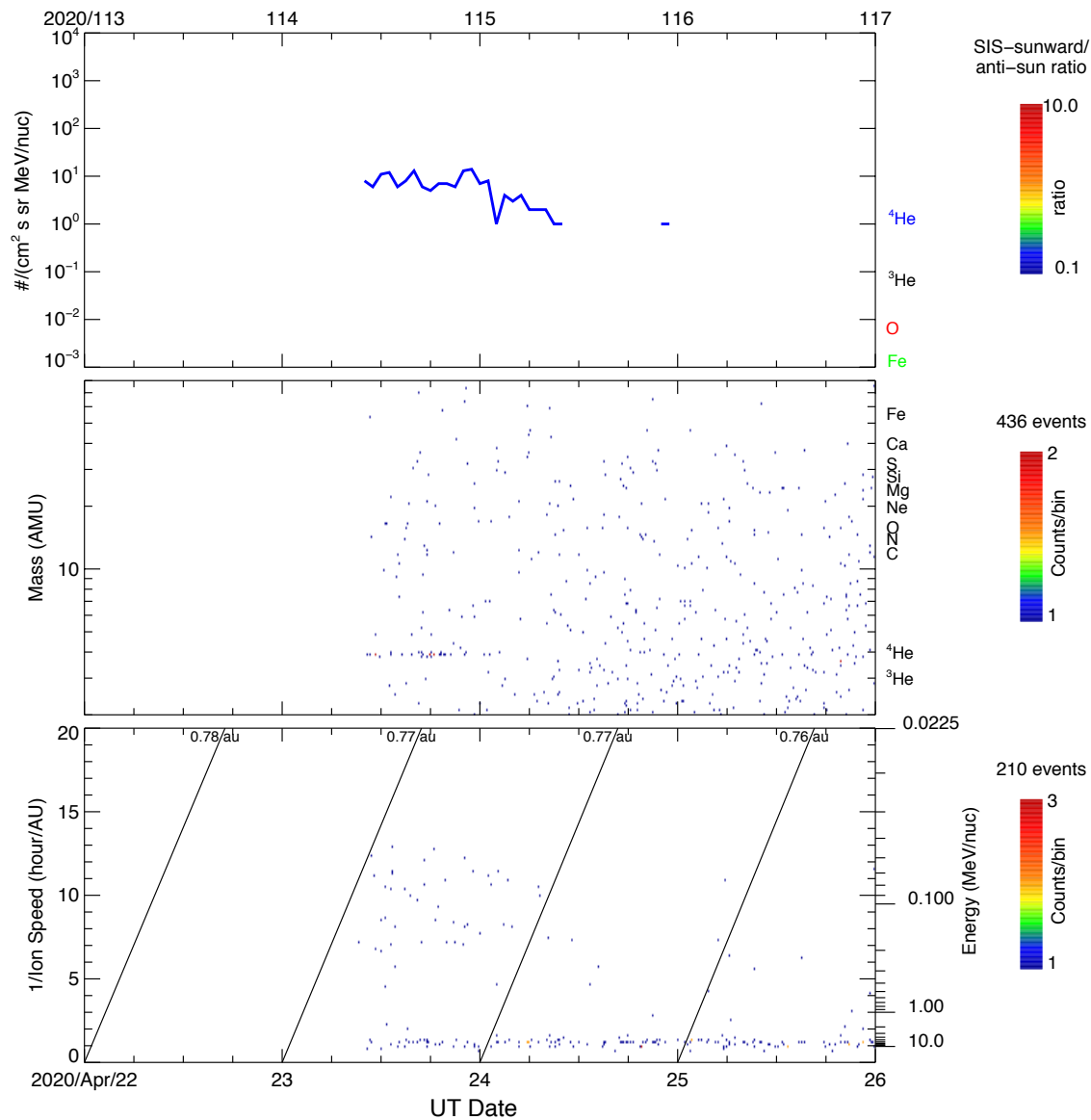
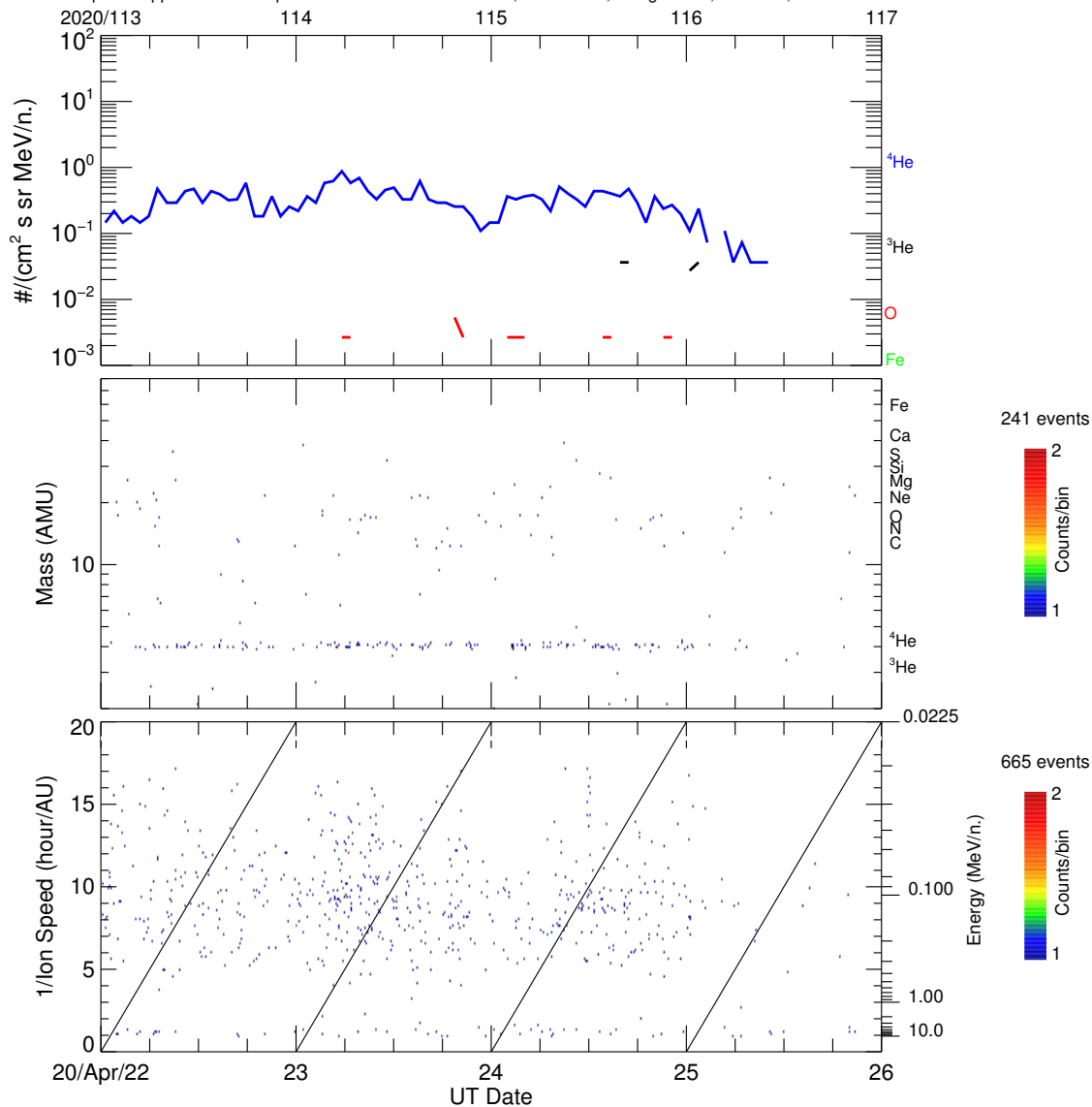
Similar plot planned
For SO/EPD/SIS

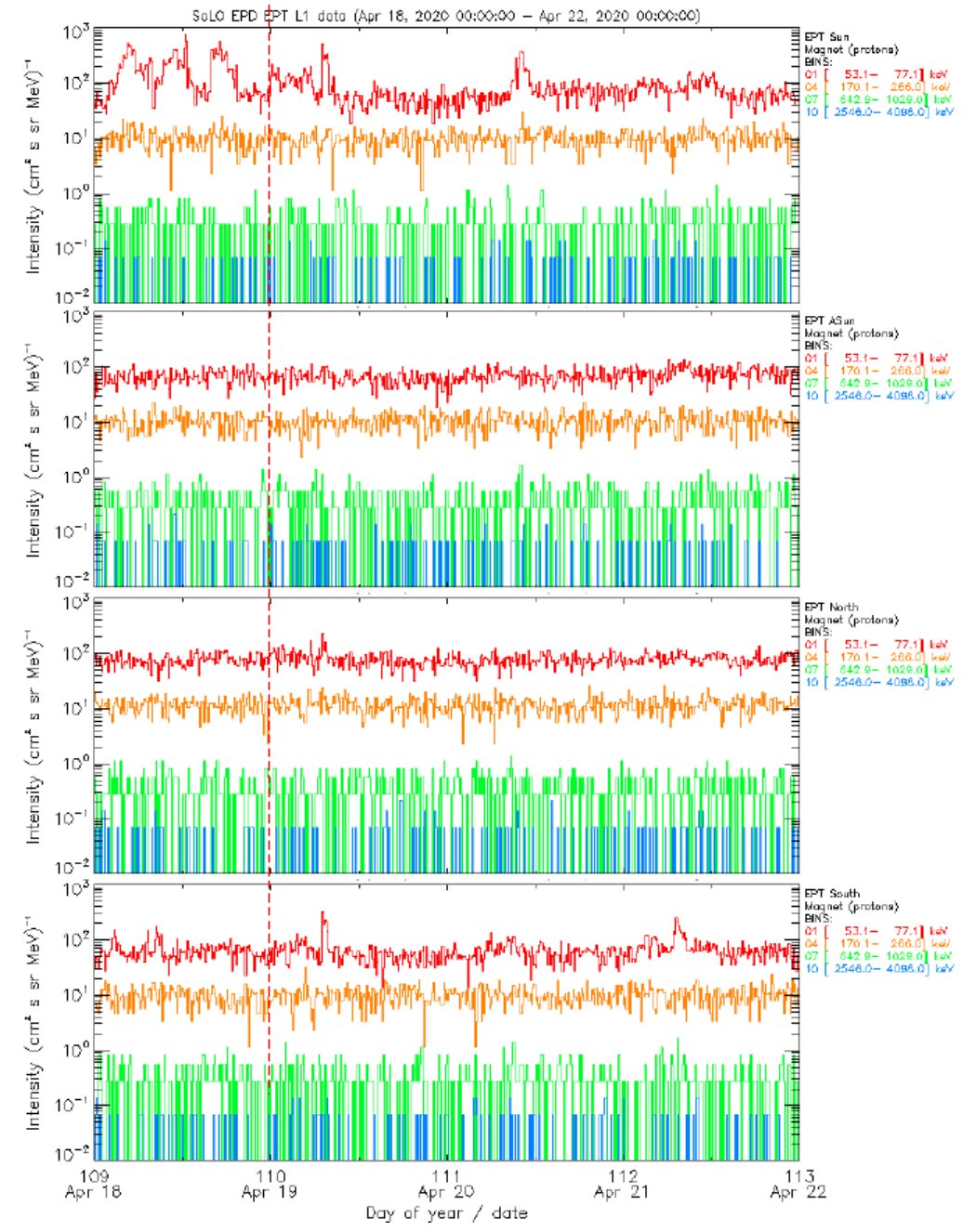
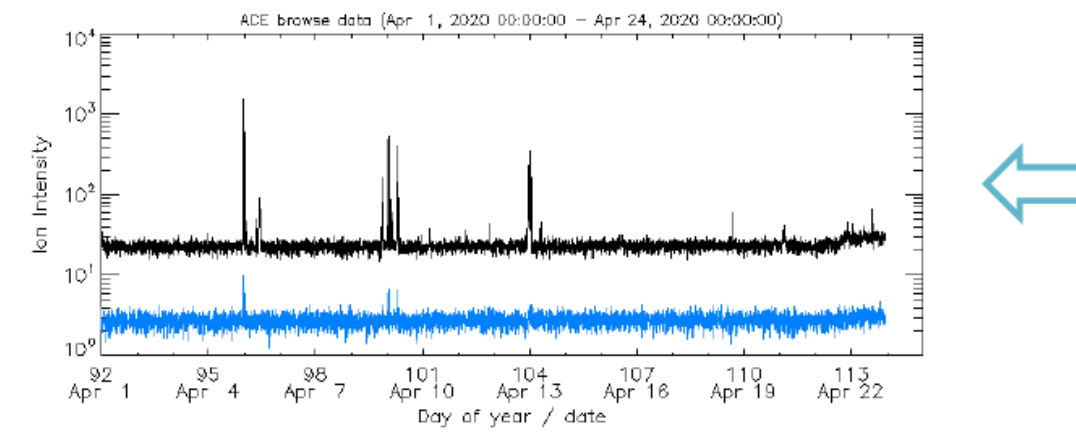
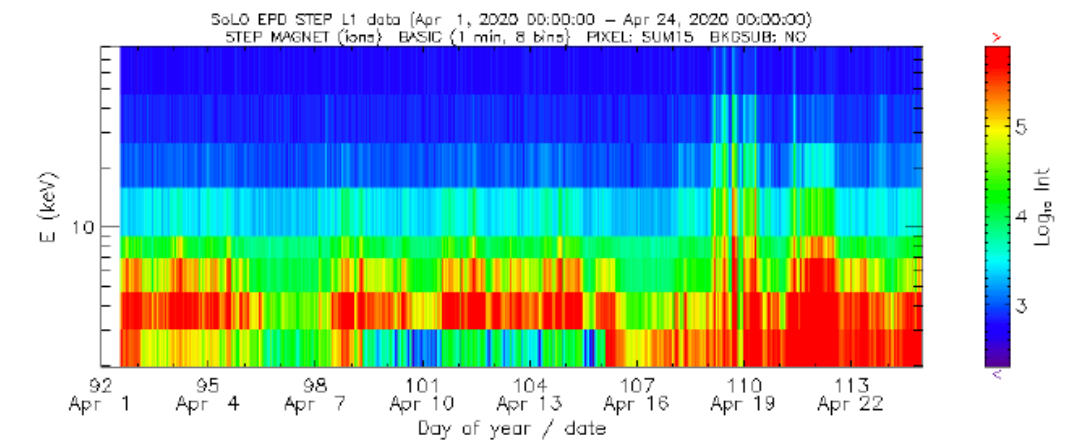
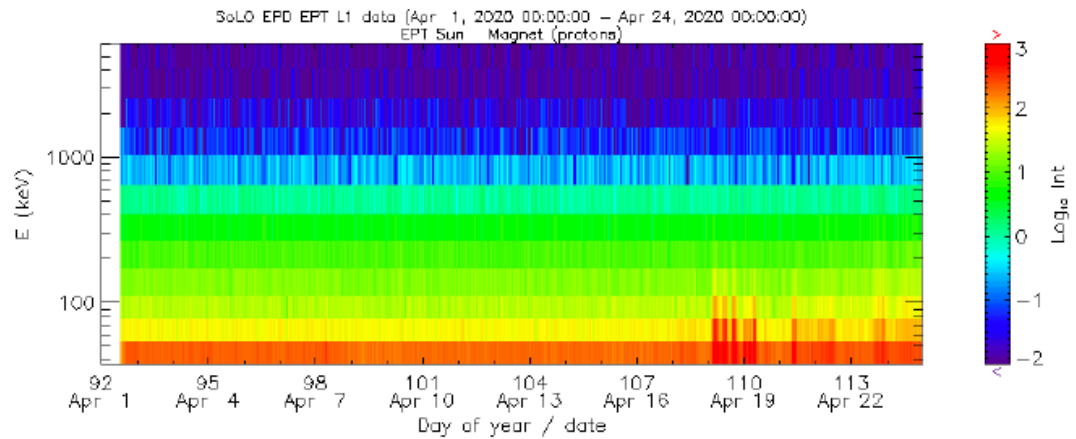
ACE/ULEIS

SO/SIS

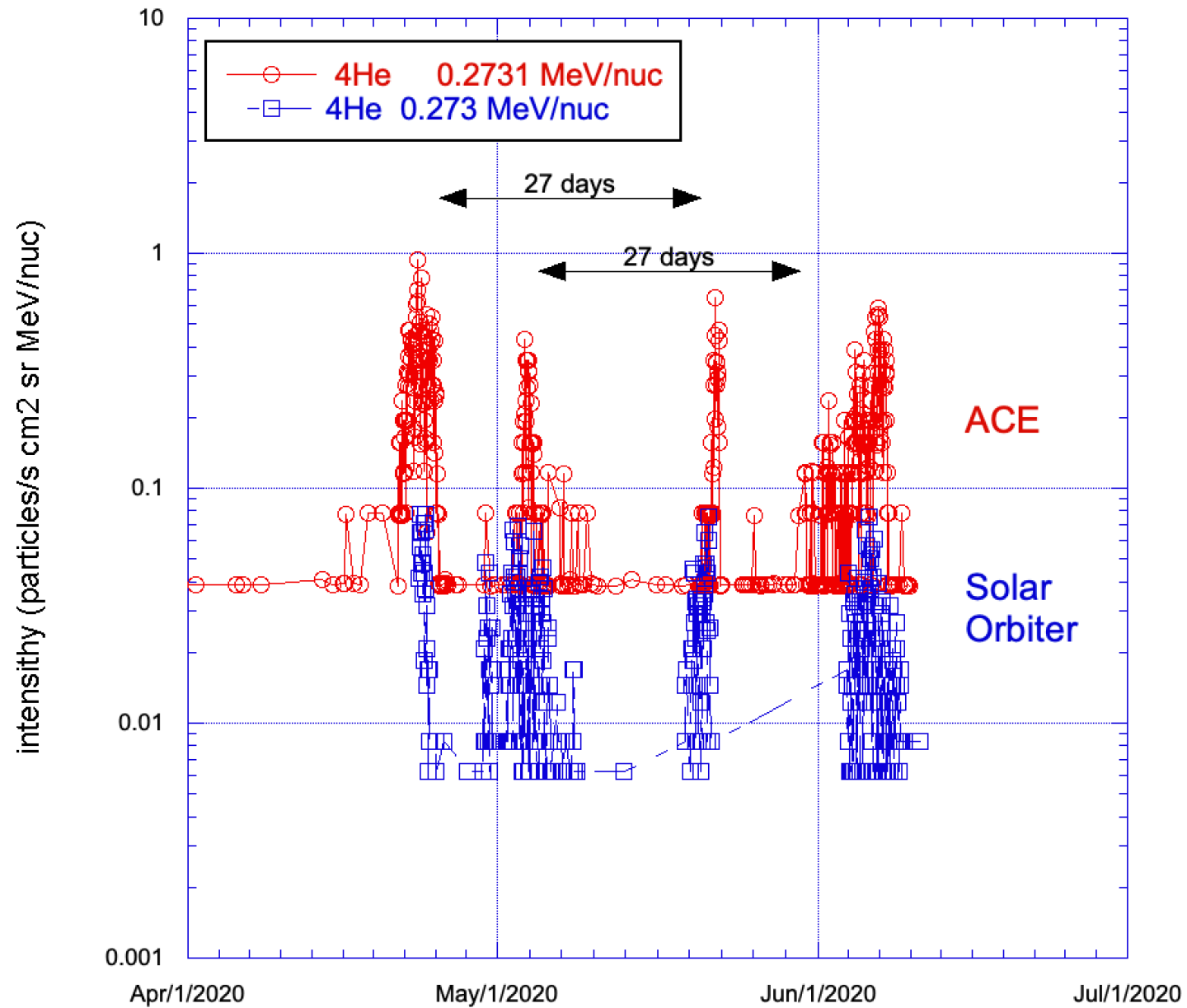
Top: 0.23–0.32 MeV/n. Intensity; Middle: Energy range = 0.4–10 MeV/n.; Bottom: Mass Range = 10–70 AMU
 Top panel: yellow shading indicates approximate period of instrument saturation
 Bottom panel: approximate door position indicator: none = 100%, blue = 26%, orange = 6%, red = 1%, black = 0%

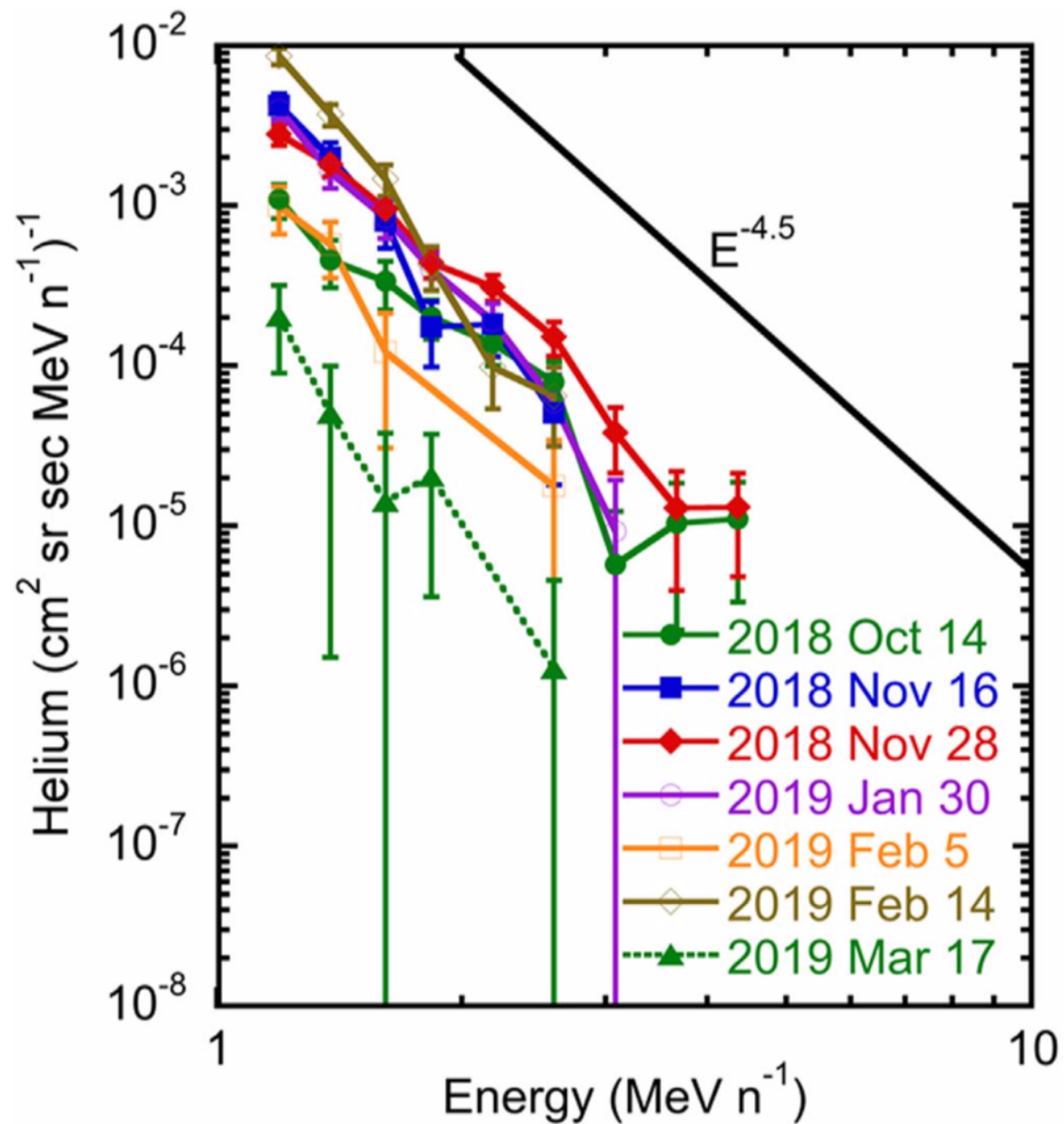
Top: 0.23–0.32 MeV/n. Intensity; Middle: Energy range = 0.4–10 MeV/n.; Bottom: Mass Range = 10–70 AMU
 Top panel: yellow shading indicates approximate period of instrument saturation
 Bottom panel: approx door position (upper = sunward): none = 100%, blue = 26%, orange = 6%, red = 1%, black = 0%





1 AU vs Solar Orbiter CIR intensities





PSP CIR He spectra

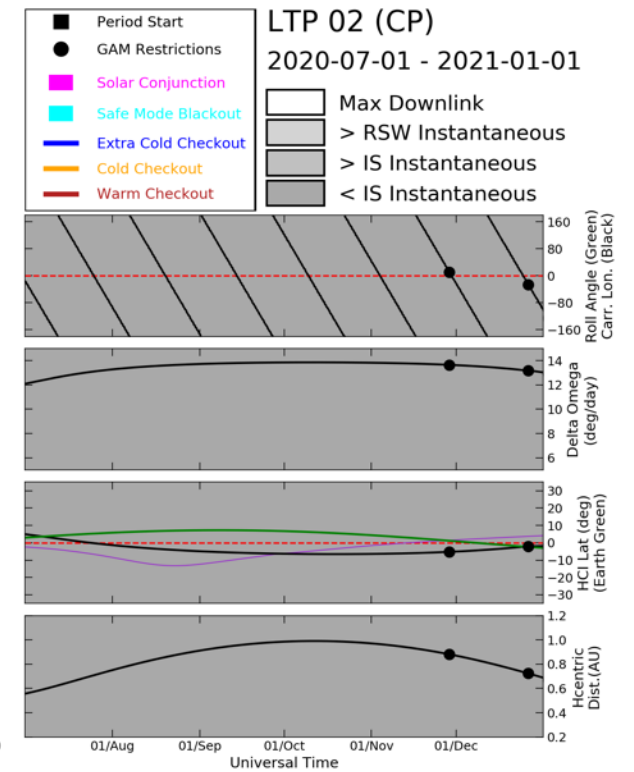
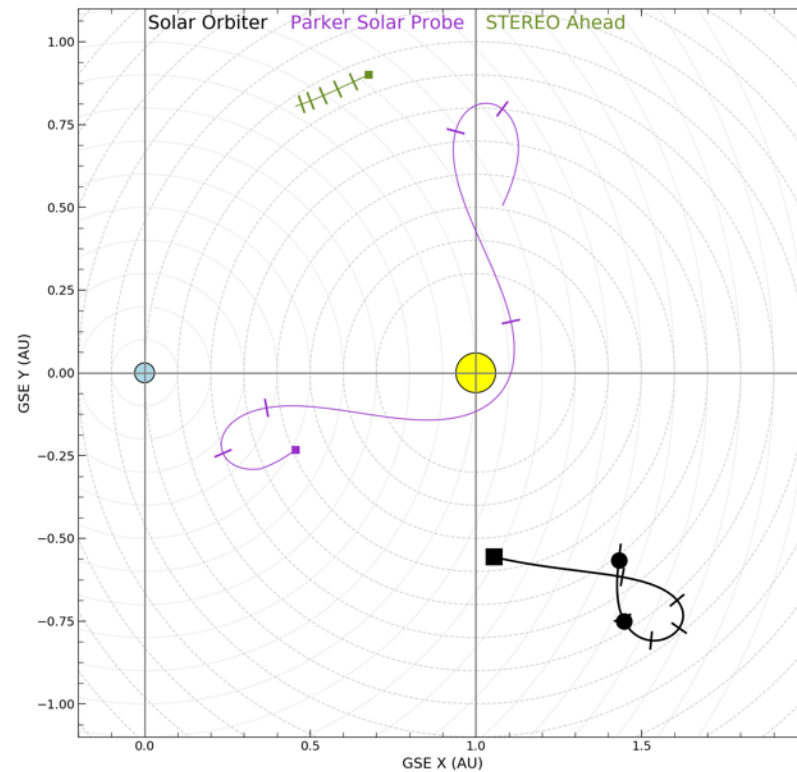
Cohen et al,
ApJ. V246, 2020.

Figure 5. Time-integrated helium spectra (from the average of LETA and LETB He spectra) for all seven events. A line corresponding to an $E^{-4.5}$ power law is shown for reference.

Solar Orbiter Current Status



- The mission just transitioned from Commissioning to Cruise Phase operation
- All in-situ instruments will be taking continuous measurement
- Remote Checkout Windows
 - June 2020
 - Feb, Mar, Sep 2021
- Venus Gravity Assist #1
 - Dec 2020
- Go Solar Orbiter !





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